EASA	AIRWORTHINESS DIRECTIVE
	AD No.: 2006–0298R1 **Date: 27 April 2009** Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.

This AD is issued in accordance with EC 1702/2003, Part 21A.3B. In accordance with EC 2042/2003 Annex I, Part M.A.301, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [EC 2042/2003 Annex I, Part M.A.303] or agreed with the Authority of the State of Registry [EC 216/2008, Article 14(4) exemption].

Type Approval Holder's Name :	**Type/Model designation(s) :**
AIRBUS	A340-200/-300 aeroplanes

TCDS Number :	EASA.A.015
Foreign AD :	None
Revision :	This AD revises EASA AD 2006-0298 dated 09 October 2006.

ATA 78	**Exhaust - Thrust Reverser Outer Fixed Structure - Modification**
Manufacturer(s):	AIRBUS (formerly AIRBUS INDUSTRIE)
Applicability:	AIRBUS A340 models -211, -212, -213, -311, -312 and -313 aeroplanes, all serial numbers, if equipped with ROHR thrust reversers serial numbers: 119 to 0382001, 0411001 to 0678001, 1035001 to 1298001 (AIRCELLE serial numbers: 3060 to 3190, 3205 to 3340 and 3525 to 3660).
Reason:	During the application of the inner fixed structure preventive mandatory modification, the manufacturer CFM International discovered an Outer Fixed Structure (OFS) panel disbonding due to an adhesive failure. In case of Fan Blade Out, an OFS disbonding, if not corrected, can lead to the in flight loss of Common Nozzle Assembly and could cause damage to the aircraft and/or hazard to persons on the ground. The aim of this AD is to replace the current DURACORE OFS panels by HEXCEL or PAA panels and to replace the 6 O'clock latch fitting for some thrust reversers depending on the total accumulated Flight Cycles (FC) at time of panel replacement. This AD has been revised to alleviate the AD applicability and to put it in line with the effectivity of AIRBUS Service Bulletin A340-78-4032 Revision 2.
Effective Date:	23 October 2006

Required action(s) and Compliance Time(s):	The following measures are rendered mandatory from the effective date of this AD: 1. For all affected thrust reversers, before the thrust reverser has accumulated 12 800 total FC since new, replace the Outer Fixed Structure panel on the left and the right half thrust reverser in accordance with the instructions of AIRBUS SB A340-78-4032 Revision 02. 2. If the thrust reverser has accumulated more than 11 600 total FC since new, at the time of OFS panel replacement as per the requirement of paragraph 1. above, replace the 6 o'clock latch fitting simultaneously in accordance with the instructions of AIRBUS SB A340-78-4032 Revision 02. 3. If AIRBUS SB A340-78-4032 has been embodied at original issue before thrust reversers accumulated 11 600 total FC since new, no further action is required to be compliant with this AD. 4. If AIRBUS SB A340-78-4032 has been embodied at original issue after thrust reversers have accumulated 11 600 total FC since new, the 6 o'clock latch fitting must be replaced in accordance with the instructions of ROHR SB RA34078-86 before thrust reversers accumulate 12 800 total FC, as required by AIRBUS SB A340-78-4032 Revision 02 and ROHR SB RA34078-75 Rev.1. 5. The accomplishment of the requirements of paragraph 1. and 2. of this AD in accordance with AIRBUS SB A340-78-4032 Revision 01 is acceptable to comply with the requirements of this AD.
Ref. Publications:	AIRBUS Service Bulletin (SB) A340-78-4032 Revision 02; ROHR SB RA34078-86; ROHR SB RA34078-75 Rev.1. The use of later approved revisions of these documents is acceptable for compliance with the requirements of this AD.
Remarks:	1. If requested and appropriately substantiated, EASA can accept Alternative Methods of Compliance for this AD. 2. This original issue of this AD was posted as PAD 06-205 for consultation on 8 August 2006 with a comment period until 23 August 2006. The Comment Response Document can be found at http://ad.easa.europa.eu/ 3. Enquiries regarding this AD should be referred to the Airworthiness Directives, Safety Management & Research Section, Certification Directorate, EASA. E-mail: ADs@easa.europa.eu. 4. For any question concerning the technical content of the requirements in this AD, please contact: AIRBUS SAS – Airworthiness Office – EAL. Fax: + 33 5 61 93 45 80 or + 33 5 61 93 44 51 E-mail: airworthiness.A330-A340@airbus.com.