EASA	AIRWORTHINESS DIRECTIVE			
X	AD No.: 2006 - 0322			
×.	Date: 18 October 2006			
		ness Directive applies, except in accordance with the rwise agreed with the Authority of the State of Registry.		
Type Approval He	older's Name :	Type/Model designation(s) :		
AIRBUS SAS		A330 and A340-200/-300 aircraft		
TCDS Number : EAS	SA A.004 & EASA A.015			
Foreign AD : none	Foreign AD : none			
Supersedure : none	Supersedure : none			
ATA 28	Fuel - Fuel tanks – P Modification / Install	revention against fuel explosion risks – ation		
Manufacturer:	AIRBUS (formerly AIRBU	SINDUSTRIE)		
Applicability:	numbers. Note : Applicability for eac	and A340-300 aircraft, all certified models, all serial h of the actions rendered mandatory is defined in on of this Airworthiness Directive (AD)		
Reason:		nich occurred to a Boeing 747-131 (flight TWA800), 88 (Special Federal Aviation Regulation 88).		
	04/00/02/07/03-L024 of F	02/07/01-L296 of March 4th, 2002 and ebruary 3rd, 2003 the JAA recommended to the ies (NAA) the application of a similar regulation.		
	passenger transport aircra	is to require all holders of type certificates for aft certified after January 1st, 1958 with a capacity e, or a payload of 3,402 kg or more, to carry out a explosion hazards.		
	Consequently, the following	ng measures are rendered mandatory:		
	fuel resistant remov	white P-clips by blue P-clips which are more re the risks of fuel quantity indicator (FQI) and stem (FLSS) harnesses chafing against the P-clip,		

	<ul> <li>modification of electrical bonding of equipment installed in fuel tanks in order to re-establish the conformity with the design definition by introducing additional bonding leads, electrical bonding points and electrical bonding of a support bracket for a diffuser assembly installed between Rib 1 and Rib 2 on the stringers of the No.1 bottom skin panel,</li> <li>modification of bonding points, installation of additional bonding leads and other modifications of the Additional Center Tank (ACT),</li> <li>modification to increase the distance between metallic parts on the THS Trim Tank,</li> <li>installation of a bonding lead between the bonding tags on the Jettison valve actuator and drive assembly.</li> </ul>
Effective Date:	02 November 2006.
Compliance:	Unless already accomplished, The following actions are rendered mandatory from the effective date of this AD: Action n°1 applicable to: AIRBUS A330, A340-200 and A340-300 aircraft all certified models, all serial numbers except for aircraft on which AIRBUS modification No. 47634 has been embodied in production to the p-clips in the wings and center lue tanks, and if necessary apply the corrective actions, in accordance with the instructions of Service Bulletin (SB) A330-28-3092 Revision 01 or SB A340-28-4107 Revision 01. AIRBUS A330, A340-200 and A340-300 aircraft, all certified models, all serial numbers except for aircraft on which the four AIRBUS modifications No. 47634 has been embodied in production of the p-clips in the wings and center lue tanks, and if necessary apply the corrective actions, in accordance with the instructions of Service Bulletin (SB) A330-28-3092 Revision 01 or SB A340-28-4107 Revision 01. Action nº2 applicable to: AIRBUS A330, A340-200 and A340-300 aircraft, all certified models, all serial numbers except for aircraft on which the four AIRBUS modifications No. 49136 and 49630 and 51825 and 55118 have been embodied in production or modified in-service in accordance with both SB A330-28-3082 and SB A330-28-3101 or both SB A340-28-4097 and SB A340-28-4018: Not later than December 31st, 2009, modify the electrical bonding of equipment installed in fuel tanks in accordance with the instructions of SB A330-28-3082 and SB A330-28-3101 or SB A340-28-4097 and SB A340-28-4017 or 44002/SB A340-200 and A340-300 aircraft, all certified models, all serial numbers, which have the AIRBUS modification 42612/SB A340-28-4047 or 44002/SB A340-28-4066 or 44005/SB A340-28-4067 embodied in production/in-service (installation of ACT) except aircraft modified by SB A340-28-4078 in service:

	Not later than December 31st, 2009, modify the electrical bonding in the ACT in accordance with the instructions of SB A340-28-4078.         Reminder:       It's the responsibility of the operator to ensure that any spare (ACT) that could be installed on the aircraft does not put in to question the compliance of the aircraft with the requirements of this AD.				
	Action n°4 applicable to:				
	<ul> <li>AIRBUS A330-300 aircraft, -301, -321, -322, -341, -342 models, all serial numbers except for aircraft on which AIRBUS modification No. 44252 has been embodied in production or modified in-service in accordance with AIRBUS SB A330-55-3016:</li> </ul>				
	- AIRBUS A340-200 and A340-300 aircraft, all certified models, all serial numbers, except for aircraft on which AIRBUS modification No. 44252 has been embodied in production or modified in-service in accordance with AIRBUS SB A340-55-4017:				
	Not later than December 31st, 2009, increase the distance between metallic parts on the THS Trim Tank in accordance with the instructions of SB A330-55-3016 or SB A340-55-4017.				
	Action n°5 applicable to: AIRBUS A340-200 and A340-300 aircraft, all certified models, all serial numbers, except for aircraft which have the modification 46142 embodied in production or modified in-service in accordance with AIRBUS SB A340-28- 4073.				
	Not later than December 31st, 2009, install a bonding lead between the bonding tags on the Jettison valve actuator and drive assembly in accordance with the instructions of SB A340-28-4073.				
Ref. Publications:	AIRBUS Service Bulletin A330-28-3092 Revision 01				
	AIRBUS Service Bulletin A340-28-4107 Revision 01				
	AIRBUS Service Bulletin A330-28-3082				
	AIRBUS Service Bulletin A330-28-3101				
	AIRBUS Service Bulletin A340-28-4097				
	AIRBUS Service Bulletin A340-28-4118				
	AIRBUS Service Bulletin A340-28-4078				
	AIRBUS Service Bulletin A330-55-3016				
	AIRBUS Service Bulletin A340-55-4017				
	AIRBUS Service Bulletin A340-28-4073				
	or later approved revisions.				

Remarks :	1.	If requested and appropriately substantiated the responsible EASA manager for the related product has the authority to accept Alternative Methods of Compliance (AMOCs) for this AD.
	2.	This AD was posted as PAD 06-198 for consultation on 7 August 2006 with a comment period until 7 September 2006. The Comment Response Document can be found at <u>http://ad.easa.europa.eu</u>
	3.	Enquiries regarding this Airworthiness Directive should be referred to Mr. M. Capaccio, Airworthiness Directive Focal Point - Certification Directorate, EASA. E-mail: <u>ADs@easa.europa.eu</u>
	4.	For any question concerning the technical content of the requirements in this AD, please contact AIRBUS SAS – Airworthiness Office - EAL Fax: +33 5 61 93 45 80.

