EASA EMERGENCY AIRWORTHINESS DIRECTIVE AD No: 2007 - 0061-E Date: 07 March 2007 No person may operate an aircraft to which an Airworthiness Directive applies, except in ccordanc of the State of requirements of that Airworthiness Directive unless otherwise agreed with the Authorit Type Approval Holder's Name: Type/Model design ti/ (s): **TURBOMECA** TURMO tu Joshan ngil TCDS Number: France M8 Foreign AD: not applicable Supersedure: 2006-0240-E Engine - Flamble I brication pipe between tank and cooler and **ATA 72** flexible lub cation between intermediate bearing and oil pump ispe on **IRB ₫ECA** Manufacturer(s): Airworthiness Directive applies to all TURMO IV turboshaft engines rted with an oil pipe P/N 0 249 92 813 0 or 0 249 92 916 0 or 249 92 808 0. Applicability: These engines are known to be installed in, but are not limited to, Eurocopter SA 330 F, G or J PUMA helicopters. Several occurrences of deterioration of: the oil pipe P/N 0 249 92 813 0 installed on TURMO III C4 between the oil cooler and the tank; the oil pipe P/N 0 249 92 808 0 installed on TURMO III C4 and TURMO IV C between the intermediate bearing and the oil pump; have been reported to Turbomeca. Most, but not all, reported occurrences Reason: concerned military models. Such deterioration and oil leakage can lead to commanded engine in flight shut down. The cause the pipe deterioration is linked to a manufacturing process change applied by the manufacturer of the pipes in 2003.

This Airworthiness Directive applies to oil pipes P/N 0 249 92 813 0,

	
	0 249 92 916 0 and 0 249 92 808 0, which are the same as, or similar to, the deteriorated pipes in military models.
	The present update introduces inspection of the pipe 0 249 92 808 0, for which two occurrences of deterioration were recently reported.
Effective Date:	09 March 2007
Compliance:	 1- For engines equipped with oil pipe P/N 0 249 92 813 0 or 0 249 92 916 0 or 0 249 92 808 0, unless already performed, the following at the are mandatory, from the effective date of this AD: Before next flight: - Identify the vulcanisation batch of the oil pipe. - If the pipe was vulcanised in the 2nd quarter of 2003, - Inspect the oil filter for black pancies which could come from internal deterioration of the pite. - Inspect the engine bay for possine oil leadage from the pipe. - Perform a boroscops dispection of the pipe. 2- If both engines installed on a liver heliconter are equipped with pipes P/N 0 249 92 813 0 or 10 9 92 16 0 yearch where vulcanised after the first quarter of 2003 replace one of the two pipes before the next flight. 3- After 25 additional flight hours, depeat the inspections required by paragraph 1 NOTE: inspection for one obefore and after flight, and inspection of the oil filter every 25 heres, as defined in the Engine Maintenance Manual, remain applicable. The observed bure is defined by the referenced TURBOMECA in the data of the pipe of t
Ref. Publication	T RBOMECA Mandatory Service Bulletin A249 72 0802 Update No. 2 or ater approved revisions. TURBOMECA Maintenance Manual 279 02 931.
Remarks :	If requested and appropriately substantiated, the responsible EASA manager for the related product has the authority to accept Alternative Methods of Compliance (AMOC) for this AD. The refets acceptant has resulted in a decision not to implement the
	 The safety assessment has resulted in a decision not to implement the full consultation process but to proceed immediately with publication and notification. Enquiries regarding this AD should be addressed to Mr. M. Capaccio,
	AD Focal Point, Certification Directorate, EASA. E mail: Ads@easa.europa.eu
	For any questions concerning the technical content of the requirements in this AD, please contact:

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