


EASA	AIRWORTHINESS DIRECTIVE	
	<p style="text-align: center;">AD No.: 2007 - 0125</p> <p style="text-align: center;">Date: 04 May 2007</p>	
<p>No person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of that Airworthiness Directive unless otherwise agreed with the Authority of the State of Registry.</p>		
<p>Type Approval Holder's Name : AIRBUS</p>	<p>Type/Model designation(s) : A330, A340-200/-300 aircraft</p>	
<p>TCDS Number: EASA A.004, EASA A.015</p>		
<p>Foreign AD: Not applicable</p>		
<p>Supersedure: Not applicable</p>		
ATA 53	Fuselage – Longitudinal joint at Stringer 28 RH between FR18 and FR19- Inspection	
<p>Manufacturer(s):</p>	<p>AIRBUS (formerly AIRBUS INDUSTRIE)</p>	
<p>Applicability:</p>	<p>AIRBUS aircraft A330-200, A330-300, A340-200 and A340-300 series, all certified models, all serial numbers (MSN) up to MSN 0402 included, except MSN 051.</p>	
<p>Reason:</p>	<p>During ground inspection of an A340-311 aircraft, it has been discovered that 5 fasteners were missing between Frame (FR) 18 and FR19 on longitudinal joint at stringer 28RH.</p> <p>Further investigations have revealed that the missing fasteners have not been installed in production due to incorrect production instructions.</p> <p>If not corrected, this situation could affect the structural integrity of the aircraft in the area of stringer 28 between FR18 and FR19 at longitudinal joint.</p> <p>In order to re-establish the structural strength of the aircraft, this Airworthiness Directive (AD) renders mandatory the inspection of the longitudinal joint at stringer 28 RH between FR18 and FR19.</p>	
<p>Effective Date:</p>	<p>18 May 2007</p>	

Compliance:	<p>Unless already accomplished,</p> <p>Before accumulation of 14000 Flight Cycles (FC) from the first flight of the aircraft or within 1500 FC following the effective date of this AD, whichever occurs later:</p> <p>in accordance with instructions defined in AIRBUS Service Bulletin (SB) A330-53-3170 or SB A340-53-4175, perform a Detailed Visual Inspection (DVI) of the longitudinal joint at stringer 28 RH between FR18 and FR19 and if necessary, apply the associated corrective actions.</p>
Ref. Publications:	<p>AIRBUS Service Bulletin A330-53-3170;</p> <p>AIRBUS Service Bulletin A340-53-4175</p> <p>or later approved revisions of these documents.</p>
Remarks :	<ol style="list-style-type: none"> 1. If requested and appropriately substantiated the responsible EASA manager for the related product has the authority to accept Alternative Methods of Compliance (AMOCs) for this AD. 2. This AD was posted as PAD 07-051 for consultation on 30 March 2007 with a comment period until 13 April 2007. No comments were received during the consultation period. 3. Enquiries regarding this Airworthiness Directive should be referred to the AD Focal Point - Certification Directorate, EASA. E-mail: ADs@easa.europa.eu . 4. For any question concerning the technical content of the requirements in this AD, please contact: AIRBUS SAS – Airworthiness Office - EAL - Fax : + 33 5 61 93 45 80 E- mail : airworthiness.A330-A340@airbus.com.