


EASA	AIRWORTHINESS DIRECTIVE	
	<p style="text-align: center;"><b>AD No : 2007-0157</b></p> <p style="text-align: center;"><b>Date: 04 June 2007</b></p>	
<p>No person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of that Airworthiness Directive unless otherwise agreed with the Authority of the State of Registry.</p>		
<p><b>Type Approval Holder's Name :</b> AIRBUS SAS</p>	<p><b>Type/Model designation(s) :</b> Aircraft A310 and A300-600</p>	
<p>TCDS Number: France No. 145</p>		
<p>Foreign AD: Not applicable.</p>		
<p>Supersedure: This AD supersedes DGAC France AD (CN) 86-102-74(B)</p>		
<p><b>ATA 53</b></p>	<p><b>Fuselage – Rear/Aft Pressure Bulkhead Attachment Angles – Modification and Inspection Program</b></p>	
<p>Manufacturer(s):</p>	<p>AIRBUS (formerly AIRBUS INDUSTRIE)</p>	
<p>Applicability:</p>	<p>AIRBUS A310-203, A310-222, A300B4-620, A300B4-622 and A300C4-620 aircraft models, except aircraft on which AIRBUS modification N. 05526 has been incorporated during production.</p>	
<p>Reason:</p>	<p>DGAC France AD 86-102-74(B) was issued to prevent development of damage, which was discovered during fatigue test in the attachment angles of the rear pressure bulkhead (fuselage frame 80/82).</p> <p>Following the life extension activities linked to the A310 program, the interval of inspection for A310-200 aircraft series has been reduced from 12,000 flight cycles (FC) to 9,000 FC.</p> <p>Consequently, this new AD, which supersedes DGAC France AD 86-102-74(B), retains the former modification requirement of DGAC AD and requires the accomplishment of the repetitive inspections and, if necessary, of associated corrective actions, at a reduced interval for A310 aircraft. The requirements applicable to A300-600 aircraft remain unchanged.</p>	
<p>Effective Date:</p>	<p>18 June 2007</p>	
<p>Compliance:</p>	<p>Compliance with the following is hereby rendered mandatory, unless already accomplished with any Service Bulletins revisions as listed in paragraph Ref. Publications of this AD (only for paragraphs 1 and 2.1):</p>	

	<p><b>1) <u>Modification</u></b></p> <p>Before the aircraft reaches 12,000 FC since first flight, embody the modification using the instructions of AIRBUS SB A300-53-6006 revision 03 or SB A310-53-2025 revision 06, as applicable.</p> <p><b>2) <u>Repetitive inspections and Corrective actions</u></b></p> <p>Applicable to aircraft on which AIRBUS SB A300-53-6006 or A310-53-2025 has been embodied between 6,000 FC and 12,000 FC since first flight:</p> <p><u>Note:</u> when AIRBUS SB A300-53-6006 or A310-53-2025 is embodied prior to 6,000FC, normal inspection program in accordance with A300-600 ALI tasks 531919-01-1A &amp; 531919-02-1A or A310 ALI task 531919-01-1A, as applicable, shall apply.</p> <p><b>2.1. <u>Before</u> embodiment of the modification:</b> Inspect critical area using the accomplishment instructions of ISB A300-53-6005 revision 03 or SB A310-53-2024 revision 05.</p> <p><b>2.2. <u>After</u> the embodiment of the modification:</b></p> <p>a) <u>For A300-600 aircraft</u></p> <p>Repeat inspection before the aircraft has reached 18,000 FC since first flight, and then at intervals not exceeding 12,000 FC in accordance with SB A300-53-6005 revision 03.</p> <p>b) <u>For A310 aircraft</u></p> <p>Repeat this inspection before the aircraft has reached 18,000 FC since first flight, and then at intervals not exceeding 9,000 FC in accordance with SB A310-53-2024 revision 05.</p> <p>For A310 aircraft which, on the effective date of this AD, are approaching or have exceeded 9,000 FC since the last inspection performed in accordance with DGAC France AD 86-102-74 requirements, inspect within the next 1,500 FC after the effective date of this AD, provided it does not result in exceeding previous interval of 12,000FC.</p>
Ref. Publications:	<p>AIRBUS Service Bulletins A300-53-6005 original issue, revision 1, revision 2 or revision 3; A300-53-6006 original issue, revision 1, revision 2 or revision 3; A310-53-2024 original issue, revision 1, revision 2, revision 3, revision 4 or revision 5; A310-53-2025 original issue, revision 1, revision 2, revision 3, revision 4, revision 5 or revision 6; or any later approved revisions of these documents.</p>
Remarks :	<ol style="list-style-type: none"> <li>1. If requested and appropriately substantiated the responsible EASA manager for the related product has the authority to accept Alternative Methods of Compliance (AMOCs) for this AD.</li> <li>2. This AD was posted on 21 March 2007 as PAD 07-047 for consultation until 18 April 2007. The Comment Response Document (CRD) can be found at <a href="http://ad.easa.europa.eu/">http://ad.easa.europa.eu/</a> .</li> <li>3. Enquiries regarding this AD should be referred to the AD Focal Point - Certification Directorate, EASA. E-mail: <a href="mailto:ADs@easa.europa.eu">ADs@easa.europa.eu</a> .</li> <li>4. For any question concerning the technical content of the requirements in this AD, please contact: AIRBUS SAS – EAW Airworthiness Office, telephone: + 33 5 61 93 36 96, facsimile: + 33 5 61 93 44 51.</li> </ol>