**EASA AIRWORTHINESS DIRECTIVE**

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<th>AD No.: 2007 - 0214</th>
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<td>Date: 07 August 2007</td>
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No person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of that Airworthiness Directive unless otherwise agreed with the Authority of the State of Registry.

**Type Approval Holder’s Name:** AIRBUS

**Type/Model designation(s):** A318, A319, A320 and A321

**TCDS Number:** EASA A.064

**Foreign AD:** Not applicable

**Supersedure:** DGAC F-2005-052R1, EASA approval No. 2005-3071

**ATA 25 Equipment/Furnishings – 103VU Box – Hinge Pin Stopper – Modification**

**Manufacturer(s):** AIRBUS, formerly AIRBUS INDUSTRIE


- AIRBUS modification 36115 in production or
- AIRBUS Service Bulletin (SB) A320-25-1535 in service

**Reason:** Cut wires and wires damaged by heat effect have been found in several 103VU boxes. Those electrical damages are caused by a contact between the horizontal hinge pin and adjacent electrical harness.

Investigation revealed that the pre-mod 36115 manufacturing process used to secure the pin by deforming the hinge ends or the hinge pin ends does not prevent the pin translatory motion.

To avoid harness damage and subsequent failure on the associated systems, DGAC Airworthiness Directive (AD) F-2005-052R1, dealing with the same subject, mandated the inspection of 103 VU panel hinge pin according to AIRBUS All Operators Telex (AOT) 25A1440.

In service experience has shown that AOT 25A1440 inspection procedure and its associated corrective actions were not sufficient to detect and solve hinge pin migration.
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<th><strong>Effective Date:</strong></th>
<th>21 August 2007</th>
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<td><strong>Compliance:</strong></td>
<td>Before March 31st 2009, install the hinge pin stopper on the internal door of the 103VU in accordance with the instructions given in SB A320-25-1535.</td>
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<td><strong>Ref. Publications:</strong></td>
<td>AIRBUS Service Bulletin (SB) A320 25-1535 original issue, or later approved revisions.</td>
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| **Remarks:**        | 1. If requested and appropriately substantiated the responsible EASA manager for the related product has the authority to accept Alternative Methods of Compliance (AMOCs) for this AD.  
2. This AD was posted on 13 June 2007 as PAD 07-094 for consultation until 04 July 2007. The Comment Response Document can be found at [http://ad.easa.europa.eu/](http://ad.easa.europa.eu/).  
3. Enquiries regarding this AD should be addressed to the AD Focal Point, Certification Directorate, EASA. E-mail [ADs@easa.europa.eu](mailto:ADs@easa.europa.eu).  
4. For any question concerning the technical content of the requirements in this AD, please contact AIRBUS - Fax 33 5 61 93 44 51. |