


<b>EASA</b>	<b>AIRWORTHINESS DIRECTIVE</b>	
	<p style="text-align: center;"><b>AD No : 2007- 0262</b></p> <p style="text-align: center;"><b>Date: 16 October 2007</b></p>	
<p>No person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of that Airworthiness Directive unless otherwise agreed with the Authority of the State of Registry.</p>		
<p><b>Type Approval Holder's Name:</b></p> <p>DASSAULT AVIATION</p>	<p><b>Type/Model designation(s):</b></p> <p>Fan Jet Falcon series, Mystère- Falcon 20-( )5 series, Mystère-Falcon 20-GF and Mystère-Falcon 200 aircraft</p>	
<p>TC/TCDS Number : DGAC France No. 35/103, 103bis and 103ter</p>		
<p>Foreign AD : Not applicable</p>		
<p>Supersedure : This directive supersedes EASA AD 2007-0118 released on 02 May 2007</p>		
<b>ATA 53</b>	<b>Fuselage Structure - Cockpit Windows Frames - Modification</b>	
<p>Manufacturers :</p>	<p>DASSAULT AVIATION (formerly Avions Marcel Dassault-Breguet Aviation)</p>	
<p>Applicability :</p>	<p>Fan Jet Falcon aircraft, all models, all serial numbers;  Mystère-Falcon 20-( )5 aircraft all models, all serial numbers;  Mystère-Falcon 20-GF aircraft, all serial numbers; and  Mystère-Falcon 200 aircraft, all serial numbers.</p>	
<p>Reason :</p>	<p>Cases of stress corrosion and fatigue have been identified on Dassault 'Falcon' series aircraft, resulting in cracks developing around the cockpit windows frames. This condition, if not corrected, could lead to premature failure of the affected cockpit window frames.</p> <p>Airworthiness Directive (AD) 2007-0118 was issued to require installation of removable fairings on the lower frames horizontal recesses, in order to allow an easier and more reliable inspection of the window frames, and to introduce new work cards, before they are made available by the release of the Aircraft Maintenance Manual (AMM) chapter 5-40 at revision 12.</p> <p>The present AD retains the requirements of AD 2007-0118 which is superseded and expands the applicability to include Model Mystère-Falcon 200 aircraft.</p>	
<p>Effective Date :</p>	<p>30 October 2007</p>	

Compliance :	<p>Required as indicated, unless accomplished previously:</p> <p>(1) Within the next 74 months after 16 May 2007 [the effective date of AD 2007-0118], modify the aircraft by installing the removable fairings in accordance with the instructions of Dassault Aviation Service Bulletin (SB) F20-701 or SB F200-0061, as applicable;</p> <p>(2) Before next flight after compliance with § (1) of this AD, as applicable:</p> <p>(a) Amend the applicable Fan Jet Falcon or Mystère-Falcon 20 AMM, chapter 5-40 by removing tasks 53-30-11, 53-30-12 and 53-30-13, and incorporating the following tasks:</p> <table border="1" data-bbox="516 506 1409 680"> <thead> <tr> <th>Task number</th> <th>Initial Inspection Threshold (time in-service since new)</th> <th>Subsequent Inspection Interval</th> </tr> </thead> <tbody> <tr> <td>53-30-14</td> <td>6 years</td> <td>6 years</td> </tr> <tr> <td>53-30-15</td> <td>24 years</td> <td>12 years</td> </tr> </tbody> </table> <p>and amend the aircraft maintenance programme accordingly.</p> <p>or:</p> <p>(b) Amend the Mystère-Falcon 200 AMM chapter 5-40 by incorporating the following tasks:</p> <table border="1" data-bbox="516 884 1409 1058"> <thead> <tr> <th>Task number</th> <th>Initial Inspection Threshold (time in-service since new)</th> <th>Subsequent Inspection Interval</th> </tr> </thead> <tbody> <tr> <td>721-2</td> <td>6 years</td> <td>6 years</td> </tr> <tr> <td>721-7</td> <td>24 years</td> <td>12 years</td> </tr> </tbody> </table> <p>and amend the aircraft maintenance programme accordingly.</p> <p><b>NOTE :</b> Pending publication of the respective revised chapter 5-40 introducing these changes, insertion of this AD into the AMM chapter 5-40 is considered as acceptable to comply with the requirements of § (2) of this directive.</p>	Task number	Initial Inspection Threshold (time in-service since new)	Subsequent Inspection Interval	53-30-14	6 years	6 years	53-30-15	24 years	12 years	Task number	Initial Inspection Threshold (time in-service since new)	Subsequent Inspection Interval	721-2	6 years	6 years	721-7	24 years	12 years
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Ref. Publications :	<p>Dassault Aviation SB F20-701, revision 2 or SB F200-0061 original issue.</p> <p>Any later approved revision of these documents is acceptable.</p>																		
Remarks :	<ol style="list-style-type: none"> <li>If requested and appropriately substantiated, EASA can accept Alternative Methods of Compliance for this AD.</li> <li>This AD was posted on 17 September 2007 as PAD 07-162 for consultation until 1st October 2007. The Comment Response Document can be found at <a href="http://ad.easa.europa.eu">http://ad.easa.europa.eu</a>.</li> <li>Enquiries regarding this AD should be referred to the AD Focal Point - Certification Directorate, EASA. E-mail: <a href="mailto:ADs@easa.europa.eu">ADs@easa.europa.eu</a></li> <li>For any questions concerning the technical content of the requirements in this AD, please contact your Dassault Falcon Field Service representative or Dassault Falcon Technical Assistance: <ul style="list-style-type: none"> <li>- For Europe, Middle East and Africa based operators : Hot Line : (33) 1 47 11 35 35 Fax (33)1 47 11 89 49</li> <li>- For USA, Canada and Mexico based operators : Help Desk : (1) 800-2FALCON (2325266) Fax (1)201 541 4740</li> <li>- All other areas : Help Desk : (1) 201 541 4747 Fax (1)201 541 4740</li> </ul> </li> </ol>																		