


<b>EASA</b>	<b>AIRWORTHINESS DIRECTIVE</b>	
	<p><b>AD No.: 2007- 0296</b></p> <p><b>Date: 07 December 2007</b></p>	
<p>No person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of that Airworthiness Directive unless otherwise agreed with the Authority of the State of Registry.</p>		
<p><b>Type Approval Holder's Name :</b> APEX AIRCRAFT</p>	<p><b>Type/Model designation(s):</b> CAP 10B</p>	
<p>TCDS Number: DGAC France No. 125</p>		
<p>Foreign AD: Not applicable</p>		
<p>Supersedure: None</p>		
<b>ATA 57</b>	<b>Main wing spar – Mount of the control stick base cover – Inspection / Repair</b>	
<p>Manufacturer :</p>	<p>APEX AIRCRAFT</p>	
<p>Applicability :</p>	<p>CAP 10B airplanes from serial number (s/n) 300 to 310 inclusive and CAP 10B airplanes from s/n 1 to 40 inclusive that have been retrofitted with carbon/wood wing ref. 5702-0104048( ).</p>	
<p>Reason :</p>	<p>A case of loose bond (ungluing) of one mounting wooden block of the control stick base cover, found during the cover reinstallation, was reported to the Type Certificate Holder (TCH) and led to the issuance of the “recommended” Service Bulletin (SB) No.031004 in February 2004. Since that date, other similar occurrences have been reported. This SB in its revision 1, has therefore been reclassified “mandatory” by the TCH.</p> <p>This Airworthiness Directive (AD) mandates inspection of the mounting blocks of the control stick base cover for loose bonds and repair, as necessary.</p> <p>These actions are intended to address the identified unsafe condition so as to prevent separation of the mounting blocks from the wing spar which could result in restricted movement of the ailerons and elevators with possible partial or complete loss of controls.</p>	

Effective Date :	21 December 2007
Compliance :	<p>At the next scheduled maintenance inspection but not later than 6 months after the effective date of this AD, inspect for correct gluing the 4 mounting wooden blocks of the control stick base cover and do all applicable corrective actions as instructed in the accomplishment instructions of APEX AIRCRAFT SB No.031004 Revision 1.</p> <p><b>NOTE:</b> Inspections and corrective actions done before the effective date of this AD, in accordance with the accomplishment instructions of APEX AIRCRAFT Service Bulletin No.031004 original issue, dated 11 February 2004 are acceptable for compliance with the requirements of the present AD.</p>
Ref. Publications :	<p>APEX AIRCRAFT Service Bulletin 031004 Original Issue and Revision 1 (The use of a later approved revision of this document is acceptable for compliance with the requirements of this AD)</p>
Remarks :	<ol style="list-style-type: none"> <li>1. If requested and appropriately substantiated, EASA can accept Alternative Methods of Compliance for this AD.</li> <li>2. This AD was posted on 16 November 2007 as PAD 07-190 for consultation until 30 November 2007. No comments were received during the consultation period.</li> <li>3. Enquiries regarding this Airworthiness Directive should be referred to the AD Focal Point, Certification Directorate, EASA. E-mail: <a href="mailto:ADs@easa.europa.eu">ADs@easa.europa.eu</a> .</li> <li>4. For any questions concerning the technical content of the requirements in this AD, please contact:  <p style="margin-left: 40px;">APEX AIRCRAFT Bureau de Navigabilité 1 route de Troyes 21121 Darois FRANCE Tel.: + 33 38035 6510 Fax: + 33 38035 6515 E-mail: <a href="mailto:airworthiness@apex-aircraft.com">airworthiness@apex-aircraft.com</a></p> </li> </ol>