EASA	AIRWORTHINESS DIRECTIVE				
1	AD No.: 2008-0104				
	Date: 29 May 2008				
	Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.				
continuing airworthiness of an ai an aircraft to which an Airworthi	rcraft shall be ensured by accom ness Directive applies, except in Agency [EC 2042/2003 Annex	3B. In accordance with EC 2042/2003 Annex I, Part M.A.301, the plishing any applicable ADs. Consequently, no person may operate n accordance with the requirements of that Airworthiness Directive I, Part M.A.303] or agreed with the Authority of the State of Registry			
Type Approval Hold	ler's Name :	Type/Model designation(s) :			
Aircraft Industries a.s		LET L410UVP-E			
TCDS Number : EASA A.026					
Foreign AD : No	Foreign AD : Not applicable				
Supersedure : No	Supersedure : None				
	r				
ATA: Multiple	Introduction of Cat	oin and Aircraft Systems Safety Modifications			
Manufacturer(s):	Aircraft Industries a.s. (formerly known as Letecké Závody a.s. alias LET)				
Applicability:	All L410UVP-E aircraft				
Reason:	Early variants of the LET L410 series had certification bases that precluded them from being certificated in EU member states. On accession to the EU, these aircraft were transferred on the basis that essential safety improvements would be introduced to enable continuing operation in EU member states.				
		improvements is to bring the aircraft to a safety other similar types certificated in EU member states.			
	accident rate than othe operations. To mitigate LET-410 aeroplanes to emergency and to inco which, if not implement in safety of flight issue	the LET-410 safety records has revealed a higher er types of similar vintage undertaking similar e this, it is necessary to improve the cabin safety of the o optimize the passenger evacuation in case of orporate a series of aircraft systems modifications need, could lead to unsafe conditions that would result es. Consequently, this Airworthiness Directive (AD) at of several modifications which consist of two			
	modifications"	ergency exits installation under wings and related			
	and				

	Part B: Remaining minimum essential safety modifications		
	For details refer to the compliance block		
Effective Date:	12 June 2008		
Required action(s) and Compliance Time(s):	Before 01 July 2010, Modify the airplane in accordance with Aircraft Industries a.s. Information Bulletin IB No. L410UVP-E/230b to embody either Part A and Part B or Part B solely of the following sets of compulsory modifications as listed below:		
	Part A: Additional emergency exits installation under wings and related modifications:		
	 Modification of fuselage structure Installation of emergency exits Modification of fuselage insulation and upholstery Anti-skid carpeting on landing gear nacelles Emergency path lighting Emergency exit signs and placards Emergency lighting wiring Other related modifications 		
	Part B: Remaining minimum essential safety modifications:		
	 Modification of locking mechanism of the front emergency exit Installation of indication of closing of the front emergency exit and covers of the front baggage compartment Installation of handle at front emergency exit Installation of labels on front emergency exit Installation of temperature indicator of onboard storage batteries Installation of non-return valves in wing fuel tank filler necks and elimination of internal fuel filler necks on wing Installation of warning device of exceeding maximum operating speed V_{MO} Re-routing of rudder and aileron trim tab cables Installation of passenger seats to prevent shifting of baggage in under-seat area Installation of water collector into pressure air piping of de-icing system of tail units Modification of fuselage harness Replacement of rudder blocking device 		
	The modifications are to be implemented by Aircraft Industries a.s. although alternative methods of compliance may be accepted (see Remark 1 below).		
	FOR AIRCRAFT MODIFIED IN ACCORDANCE WITH PARTS A AND B, THERE ARE NO ADDITIONAL LIMITATIONS beyond those listed in Type Certificate Data Sheet EASA.A.026.		
	AIRCRAFT MODIFIED IN ACCORDANCE WITH PART B ONLY WILL BE LIMITED TO A MAXIMUM OF NINE PASSENGERS after 30 June 2010. (This restriction does not apply to aircraft used in parachuting operations.)		
	NOTE: These limitations are set out in Type Certificate Data Sheet EASA.A.026, taking effect on the compliance date of this AD.		
Ref. Publications:	Aircraft Industries a.s. Information Bulletin IB No. L410UVP-E/230b		

Remarks :	1.	If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.
	2.	This AD was posted on 15 February 2008 as PAD 08-022 for consultation until 14 March 2008.The Comment Response Document can be found at http://ad.easa.europa.eu .
	3.	Enquiries regarding this AD should be referred to the Airworthiness Directives, Safety Management & Research Section, Certification Directorate, EASA. E-mail <u>ADs@easa.europa.eu</u>
	4.	For any questions concerning the technical content of the requirements in this AD, please contact:
		AIRCRAFT INDUSTRIES, a.s.
		Na Záhonech 1177, 686 04 Kunovice, CZECH REPUBLIC
		Customer Support: Phone:+420-572817660 Fax:+420-572816112
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