

EASA	AIRWORTHINESS DIRECTIVE	
	<p>AD No.: 2009-0082</p> <p>Date: 07 April 2009</p> <p>Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation</p>	
<p>This AD is issued in accordance with EC 1702/2003, Part 21A.3B. In accordance with EC 2042/2003 Annex I, Part M.A.301, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD unless otherwise specified by the Agency [EC 2042/2003 Annex I, Part M.A.303] or agreed with the Authority of the State of Registry [EC 216/2008, Article 14(4) exemption].</p>		
<p>Type Approval Holder's Name :</p> <p>328 Support Services GmbH</p>	<p>Type/Model designation(s) :</p> <p>328-100 and 328-300 aeroplanes</p>	
<p>TCDS Number : EASA.A.096</p>		
<p>Foreign AD : Not applicable</p>		
<p>Supersedure : None</p>		
<p>ATA 25</p>	<p>Equipment/Furnishings – Flight Compartment Door Locking Device – Inspection</p>	
<p>Manufacturer(s):</p>	<p>Dornier Luftfahrt GmbH; Fairchild-Dornier GmbH; AvCraft Aerospace GmbH</p>	
<p>Applicability:</p>	<p>Model 328-100 aeroplanes, all serial numbers; and Model 328-300 aeroplanes, all serial numbers.</p>	
<p>Reason:</p>	<p>A recent incident has been reported with a Dornier 328-100 aeroplane, where the right-hand (RH) power lever jammed in flight-idle position during the landing roll-out. The aeroplane was stopped by excessive braking.</p> <p>The investigation by the operator revealed that the cockpit door locking device Part Number 001A252A3914012 had fallen off the RH cockpit wall and blocked the RH power/condition lever pulley/cable cluster below the door. Although the affected aeroplane had been modified, the technical investigation showed that a loose Cockpit Door Locking device could also occur on 328-100 and 328-300 aeroplanes with a standard installation.</p> <p>This condition, if not corrected, could cause interference with the engine- and/or flight control cables, possibly resulting in reduced control of the aeroplane.</p> <p>For the reasons described above, this AD requires a one-time inspection of the cockpit door locking device and the surrounding area and the reporting of all findings to the TC holder. This AD is considered to be an interim action and the retrofit of a new design may be implemented later.</p>	
<p>Effective Date:</p>	<p>21 April 2009</p>	

<p>Required Action(s) and Compliance Time(s):</p>	<p>Required as indicated, unless accomplished previously:</p> <ol style="list-style-type: none"> (1) Within the next 3 months after the effective date of this AD, accomplish a Detailed Visual Inspection of the cockpit door locking device and the surrounding area in accordance with the accomplishment instructions of 328 Support Services GmbH Service Bulletin (SB) SB-328-25-485 or SB-328J-25-235, as applicable to aeroplane model. (2) If discrepancies are found during the inspection as required by paragraph (1) of this AD, before next flight, accomplish the corrective actions in accordance with the accomplishment instructions of 328 Support Services GmbH SB-328-25-485 or SB-328J-25-235, as applicable to aeroplane model. (3) Within 30 days after the inspection as required by paragraph (1) of this AD, send an inspection report to 328 Support Services GmbH by using the Compliance Form attached to the applicable SB.
<p>Ref. Publications:</p>	<p>328 Support Services GmbH SB-328-25-485 (for 328-100 aeroplanes) and 328 Support Services GmbH SB-328J-25-235 (for 328-300 aeroplanes), both dated 28 January 2009.</p> <p>The use of later approved revisions of these documents is acceptable for compliance with the requirements of this AD.</p>
<p>Remarks :</p>	<ol style="list-style-type: none"> 1. If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD. 2. This AD was published on 16 March 2009 as PAD 09-050 for consultation until 06 April 2009. No comments were received during the consultation period. 3. Enquiries regarding this AD should be referred to the Airworthiness Directives, Safety Management & Research Section, Certification Directorate, EASA, E-mail: ADs@easa.europa.eu. 4. For any question concerning the technical content of the requirements in this AD, please contact: 328 Support Services GmbH Global Support Centre, P.O. Box 1252 D-82231 Wessling, Federal Republic of Germany Telephone: +49 8153 88111 6666, Fax: 49 8153 88111 6565 E-mail: gsc.op@328support.de.