EASA AD No.: 2009-0087-E

## EASA

## **EMERGENCY AIRWORTHINESS DIRECTIVE**

AD No.: 2009-0087-E

Date: 11 April 2009

Note: This Emergency Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation

This AD is issued in accordance with EC 1702/2003, Part 21A.3B. In accordance with EC 2042/2003 Aprex I, Part A.301, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequent on operson my operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise pecified by to Agency [EC 2042/2003 Annex I, Part M.A.303] or agreed with the Authority of the State of Registry [EC 216/203, Article 144) exercition].

[EC 2042/2003 Annex I, Part	M.A.303] or agreed with the Authority	of the State of Registry [EC 216/27 8, Article (4) exerction].
Type Approval H	older's Name :	Type/Model de gnati n(s):
EUROCOPTER		AS 332 arg 225 elications
TCDS Number :	EASA.R.002	
Foreign AD :	Not applicable	
Supersedure :	None	
ATA 63		Magnetic Plug of Main Gear Box Epicyclic uction Gear Module – Check
Manufacturer(s):	Ex-scopte of former Euro	ocopter France)
Applicability:	A 332 12 1d EC 225 LF	Phelicopters, all serial numbers.
		ness Directive (AD) is issued following the accident of registered G-REDL that occurred on the 1 <sup>st</sup> of April tland near Aberdeen.
5	accident seems to be con the MGB, the root cause of	progress and at this early stage, the cause of this nected with a degradation of the epicyclic module of of which is still to be determined. In the light of this of any contamination of the MGB is of utmost onary measure.
Reason:	For the reasons described	d above, this AD requires:
	reduction gear module maintenance has bee	ters, to check the magnetic plug of the MGB epicyclic e on a daily basis and to ensure also that correct MGE n applied in case a particle was found on the issuance of this AD; and
	reduction gear module that correct MGB main	ters, to perform maintenance of the MGB epicyclic e each time chip detection occurs and to ensure also ntenance has been applied in case a particle was magnetic plug before issuance of this AD.

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Effective Date:	11 April 2009	
	Required as indicated, unless accomplished previously:	
	1. For AS 332 L2 helicopters:	
	1.1 Before next flight after the effective date of this AD, and thereafter each day After the Last Flight of the day (ALF inspection) without exceeding 10 Flight Hours (FH) between each ALF, check the MGB epicyclic reduction gear module magnetic plug in accordance with the instructions of paragraph 2.B.1 of Eurocopter AS332 ASB No. 05.00.81 initial issue.	
	1.2 Before next flight after the effective date of this AD, check the helicopter maintenance records in order to determine if, within the law 200 H prior to the effective date of this AD, one or more particles were detected to the magnetic plug of the epicyclic reduction gear module without ensuing complete compliance with maintenance Work Care (WC) 100.007 (2.001)	
	If detection occurred and was not followed by complete in the enance (epicyclic reduction gear module not removed to eneck chip collector magnets), before further flight, disassemble at inspect the MGB epicyclic reduction gear module in accordance with the instructions of paragraph 2.B.2 of Eurocopter (15332 ASB No. 05.00.81 initial issue.	
Required Action(s) and Compliance	2. For EC 225 LP helicopters:	
Time(s):	2.1 After the effective date of the AD, each the practicles are detected on the electric magnetic plug of the MGP epicyclic reduction gear module, before further flight, disassemble and improve the MGB epicyclic reduction gear module in accordance with the estructions of paragraph 2.B.2 of Eurocopter EC221 ASB No. 05Ab unitial issue.	
	2.2 Before next flight at the effective date of this AD, check the helicopter maintenant is pords in order to determine if, within the last 200FH prior to the effective date of this, one or more particles were detected on the electric pagners the of the epicyclic reduction gear module without engling a molete disassembly and inspection of the module:	
	If detection occurred and was not followed by maintenance as defined above, before further flight, disassemble and inspect the MGB epicyclic reduction gear module in accordance with the instructions of paragraph 3 B 2 of Eurocopter EC225 ASB No. 05A017 initial issue.	
	For AS 332 L2 and EC 225 LP helicopters:	
	After the effective date of this AD, do not install on any helicopter a MGB, unless its condition has been checked in accordance with the requirements of this AD.	
Ref ablic tions:	Eurocopter AS332 ASB 05.00.81 initial issue, dated 10 April 2009.	
	Eurocopter EC225 ASB 05A017 initial issue, dated 10 April 2009.	
	The use of later approved revisions of these documents is acceptable for compliance with the requirements of this AD.	

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	1.	If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.
	2.	The safety assessment has requested not to implement the full consultation process and an immediate publication and notification.
Remarks :	3.	Enquiries regarding this AD should be referred to the Airworthiness Directives, Safety Management & Research Section, Certification Directorate, EASA. E-mail: <a href="mailto:ADs@easa.europa.eu">ADs@easa.europa.eu</a> .
	4.	For any question concerning the technical content of the requirements in this AD, please contact:  EUROCOPTER (STDI) – Aéroport de Marseille Provence 13725 Marignane Cedex, France; telephone +33 (4) 12 85 97 97; facsimile +33 (4) 85 99 66; E-mail: <a href="mailto:Directive.technical-support@eurocopter.com">Directive.technical-support@eurocopter.com</a>

