EASA	AIRWORTHINESS DIRECTIVE	
	AD No.: 2009-0108	
×.	Date: 05 May 2009	
	Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.	
continuing airworthiness of an operate an aircraft to which an	aircraft shall be ensured by ac AD applies, except in accordance	3B. In accordance with EC 2042/2003 Annex I, Part M.A.301, the complishing any applicable ADs. Consequently, no person may ce with the requirements of that AD unless otherwise specified by with the Authority of the State of Registry [EC 216/2008, Article
Type Approval Holder's Name :		Type/Model designation(s) :
AIRBUS		A340 aeroplanes
TCDS Number : EASA.A.015		
Foreign AD : Not applicable		
Supersedure :	None	
ATA 71	ATA 71 Power Plant – Forward Engine Mount Thrust Links - Inspectio	
Manufacturer(s):	AIRBUS (formerly AIRBUS INDUSTRIE)	
Applicability:	AIRBUS A340 model -211, -212, -213, -311, -312 and -313 aeroplanes, all serial numbers.	
Reason:	A review of A340 missions has demonstrated that CFM56-5C forward engine mount thrust links fitted with oversized bearing will not reach the updated link fatigue life limit of 15 500 Flight Cycles (FC) due to an increase in bore diameter.	
	failure of the affected	MM 71-21-12 Repair 1. The consequent potential thrust link would reduce the forward engine mounts nd could eventually lead to engine separation,
	Consequently, this AD	requires :
		ne link assembly to identify a possible oversized in case of finding, the application of the associated
	accomplishment of	pection of the forward engine mounts until the inspection of the link assembly for the ssible oversized bearing repair.

Effective Date:	19 May 2009	
Required action(s) and Compliance Time(s):	Required as indicated : Within 1 700 FC or 24 months from the effective date of this AD, whichever occurs first, apply the actions defined in paragraph 1. <u>OR</u> paragraph 2. of this AD:	
	<ol> <li>Unless already accomplished, inspect the link assembly for oversize bearing repair in accordance with the instructions of AIRBUS Service Bulletin (SB) A340-71-4007.</li> </ol>	
	1.1. If oversized bearings are found installed, before next flight contact Goodrich to get appropriate instructions and apply them.	
	1.2. If oversized bearings are not found installed, no further action is required by this AD.	
	2. Perform a detailed inspection of the forward engine mounts and thereafter repeat the inspection at intervals not exceeding 1 700 FC or 24 months, whichever occurs first, from the last inspection until accomplishment of the inspection defined in paragraph 1. of this AD, which must occur no later than 4 500 FC from the effective date of this AD, and apply the associated corrective actions in accordance with the instructions of AIRBUS SB A340-71-4007.	
Ref. Publications:	AIRBUS Service Bulletin A340-71-4007 at original issue.	
	The use of later approved revisions of this document is acceptable for compliance with the requirements of this AD.	
Remarks:	<ol> <li>If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.</li> </ol>	
	<ol> <li>This AD was posted on 18 March 2009 as PAD 09-054 for consultation until 17 April 2009. The Comment Response Document can be found at: <u>http://ad.easa.europa.eu/</u>.</li> </ol>	
	<ol> <li>Enquiries regarding this AD should be referred to the Airworthiness Directives, Safety Management &amp; Research Section, Certification Directorate, EASA. E-mail: <u>ADs@easa.europa.eu</u>.</li> </ol>	
	<ol> <li>For any question concerning the technical content of the requirements in this AD, please contact: AIRBUS SAS – Airworthiness Office – EAL. Fax: +33 5 61 93 45 80, E-mail: <u>airworthiness.A330-A340@airbus.com</u>.</li> </ol>	