EASA	AIRWORTHINESS DIRECTIVE	
1	AD No.: 2009-0132	
	Date: 23 June 2009)
F	tion (EC) No 216/2008 on b	rective (AD) is issued by EASA, acting in accordance with Regula- ehalf of the European Community, its Member States and of the at participate in the activities of EASA under Article 66 of that Regu-
inuing airworthiness of an air aircraft to which an Airworthin	craft shall be ensured by accomplis ess Directive applies, except in accor- gency [EC 2042/2003 Annex I, Part	. In accordance with EC 2042/2003 Annex I, Part M.A.301, the con hing any applicable ADs. Consequently, no person may operate a ordance with the requirements of that Airworthiness Directive unles M.A.303] or agreed with the Authority of the State of Registry [EC
Type Approval Ho	•	Type/Model designation(s) :
Scheibe Aircraft GmbH		Mü13E "Bergfalke" series and SF26A "Standard" sailplanes, and SF25 series and SF28A "Tandem Falke" powered-sailplanes
TCDS Numbers :	EASA A.098, A.099, A.103 ar	nd A.107
Foreign AD :	Not applicable	
Supersedure :	None	
ATA 27	Elevator and Tab - D	Prive Arm of the Mechanical Elevator Trim Tab- Inspection / Repair
Manufacturer(s):	Scheibe Aircraft GmbH, Scheibe Flugzeugbau GmbH and Sportavia Pützer GmbH & Co. KG	
Applicability:	Model Mü13E "Bergfalke" series and model SF26A "Standard" sailplanes, all serial numbers, and Model SF25 series and model SF28A "Tandem Falke" powered-sailplanes, all serial numbers.	
	arm of the mechanical e	dvised of receiving a report of looseness of the drive elevator trim tab, found during an annual inspection. ikely caused by penetrated humidity over the years.

Effective Date:	07 July 2009	
Required Action(s) and Compliance Time(s):	 Required as indicated, unless accomplished previously: (1) At the next scheduled maintenance check but no later than 12 months after the effective date of this AD, inspect the drive arm of the mechanical elevator trim tab in accordance with the initial issue of Scheibe Aircraft Service Bulletin (SB) 104-24, 232-6, 653-91 or 770-30, as applicable. If any damage is found, before further flight, repair the drive arm of the mechanical elevator trim tab in accordance with the initial issue of Scheibe Aircraft Work Instruction 104-24, 232-6, 653-91 or 770-30, as applicable. (2) Thereafter, at intervals not to exceed 12 months, repeat the inspection of the drive arm of the mechanical elevator trim tab and do all corrective actions in accordance with the initial issue of Scheibe Aircraft SB 104-24, 232-6, 653-91 or 770-30, as applicable. 	
Ref. Publications:	Scheibe Aircraft Service Bulletin 104-24, 232-6, 653-91 and 770-30, initial issue dated 23 March 2009; Scheibe Aircraft Work Instruction 104-24, 232-6, 653-91 and 770-30, initial issue dated 23 March 2009. The use of later approved revisions of these documents is acceptable for compliance with the requirements of this AD.	
Remarks :	 If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD. This AD was posted on 13 May 2009 as PAD 09-068 for consultation until 10 June 2009. No comments were received during the consultation period. Enquiries regarding this AD should be referred to the Airworthiness Directives, Safety Management & Research Section, Certification Directorate, EASA. E-mail <u>ADs@easa.europa.eu</u>. For any question concerning the technical content of the requirements in this AD, please contact: SCHEIBE AIRCRAFT GMBH / Customer Service Am Flugplatz 5, 73540 Heubach, GERMANY Telephone: + 49 (0) 7173-184286 Facsimile: + 49 (0) 7173-185587 E-mail: info@scheibe-aircraft.de 	