EASA

AIRWORTHINESS DIRECTIVE



AD No.: 2009 - 0192

Date: 28 August 2009

Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.

This AD is issued in accordance with EC 1702/2003, Part 21A.3B. In accordance with EC 2042/2003 Ang. 1, Part Ma 301, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no per in may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD unless otherwise specific by the Agency [EC 2042/2003 Annex I, Part M.A.303] or agreed with the Authority of the State of Registration [EC 218, 208, Article 14(4) exemption].

Type Approval Holder's Name:		Type/Model design tign(s):
AIRBUS		A340 aer slanes
TCDS Number:	EASA.A.015	
Foreign AD:	Not applicable	
Supersedure:	None	S
ATA 05		nance Checks – Airworthiness Limitation Amendment
Manufacturer(s):	Airb (former Airb	us Industrie)
Applicability:		nes, -211, -212, -213, -311, -312, -313, -541, -542, -642 Manufacturer Serial Numbers.
Reason:	steel forgings used to below specification li in Airbus A340 Airwo	ality tests on material, Airbus found that some 300M o manufacture certain landing gear components were mits. Adapted airworthiness limitations were introduced orthiness Limitations Section (ALS) Part 1 "Safe Life tion Items (SL ALI)" for different source route codes.
		(S3) has since been investigated and has led to another thiness limitations for affected main fittings.
	selection led to life li	ests based on a new method for the damage load mitation changes for A340-500/-600 centre landing gea ber (P/N) 50-1105236-01 and P/N 50-1105296-00.
	This AD introduces r components.	nore restrictive life limitations for several landing gear
Effective Date:	11 September 2009	
Required Action(s) and Compliance Time(s):	Incorporate in	d, from the effective date of this AD: to the approved Aircraft Maintenance Programme all anges to airworthiness limitations included in the

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	appended Table 1 and Table 2 of this AD, and						
	 Comply with any action required by the approved Aircraft Maintenance Programme following incorporation of changes introduced by this AD. 						
Ref. Publications:	AIRBUS A340 ALS Part 1 revision 03 approved under reference EASA.A.C.12282 on 29 May 2009.						
	The use of later approved revisions of this document is acceptable for compliance with the requirements of this AD.						
Remarks :	If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.						
	2. The required actions and the risk allowance have granter the istrance of a Final AD with Request for Comments, postponing the public consultation process after publication.						
	3. Enquiries regarding this AD should be referred the Airwochine's Directives, Safety Management & Research Jection Pertification Directorate, EASA. E-mail: ADs@easa.europs.cu						
	4. For any questions concerning the technical content of the requirements in this AD, please contact: AIRBUS – Airworthiness Office – E.N.; Fax: +33 5 61 93 45 80. E-mail: airworthiness.A330-A2 20@air vs.com						

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A340-200/-300 SERIES

(*) Whichever occurs first. Explanation notes are located at the bottom of the table.		LIFE LIMITATIONS (*)			L'ALIMITATION APPLA APPLA				
Nomenclature	Part Number	Flight Hours	Landings	Calendar Time	A340-2xx	Veight Val. (WV) 00x	WV¢	Other WV02x	WV05x
ATA 32-11-00 MAIN LANDIN	IG GEAR		1					L	
Main fitting (Left Hand)		91000	14035	N/A		X (1)			
	201488324 (2) (3)	90000	20910	N/A	excl. 204				
		100000	22000	N//				-312/-313 (1)	-313 (1)
		91000	14035	A		(1)			
	201488335 (2)	90000	20910		. 204 (
		100000	24400	N/A				-312/-313 (1)	-313 (1)
	50-2107001-00	75000	100	N/A				-312/-313 (1)	-313 (1)
Main fitting (Right Hand)		91000	14035	N/A		X (1)			
	201488325 (2) (3)	9000	20910		excl. 204 (1)				
		10000	20	N/A				-312/-313 (1)	-313 (1)
		91000	14 -	N/A		X (1)			
	201488336	90000	2091	N/A	excl. 204 (1)				
		100 0	24400	N/A				-312/-313 (1)	-313 (1)
	50-2108001-00	J 000	15 J	N/A				-312/-313 (1)	-313 (1)

⁽¹⁾ These limitations apply to parts in the control of the control

Table 1

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⁽²⁾ The installation of parts identified as 'S3' in lessier-between Letter A33-34 A20 dated 31-Jul-2009 is prohibited on A340-213 WV021 and A340-313 WV027 POST MOD 46651.

⁽³⁾ Rework in accordance with A RUS State Bulletin (SB) A340-32-4259 / Vendor SB A33/34-32-257 (AIRBUS Modification 56498) before accumulation of 8750 Landings introduces of the control of the control

A340-500/-600 SERIES

(*) Whichever occurs first. Explanation notes are located at the bottom of the table.		LIFE	LIMITATION	NS (*)	FE LIME YONS, PLY BILITY				
Nomenclature		Flight Hours	Landings	Calendar Time	340-5xx		A340-6xx		
	Part Number				WV00x	WV10x	WV00x	WV10x	
ATA 32-11-00 MAIN LANDING	GEAR								
Main fitting (Left Hand)		67000	7495	N/A			X (2)		
	50-1209001-03	80000	10045	N/A	Χ				
	30-1209001-03	119000	12615	N/A				X (2)	
		125000	20600			X (2)			
Main fitting (Right Hand)		67000	7495	N/A			X (2)		
	50-1210001-03	80000	1004	N/^	X (2,				
	30-1210001-03	119000	12615	/A				X (2)	
		125000	600	N/A		X (2)			
ATA 32-15-00 CENTER LAND	ING GEAR								
Main fitting		ა 40	9 _0	N/A			X (2)		
	50-1105236-01	3. 0	0020	./A	X (2)				
	30-1103230-01	3803	6520	N/A			X (3)		
		39560	6520	N/A	X (3)				
		9 970	35	N/A			X (1)		
		31250	5150	N/A	X (1)				
	296-0	4795°	7945	N/A			X (2)		
	3-1 290-00	0ر 4	7945	N/A			X (3)		
		0000	8240	N/A	X (2)				
		50000	8240	N/A	X (3)				
	1105326-00	32530	5390	N/A				X (2)	
	103320=00	40530	6810	N/A		X (2)			

⁽¹⁾ These limitations apply to particle dentified 'S1' Messier-Dowty Service Letter A33-34 A20 dated 31-Jul-2009.

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⁽²⁾ These limits apply to parts intified as 'S3' in Messier-Dowty Service Letter A33-34 A20 dated 31-Jul-2009.

⁽³⁾ These limitation oply to parts residentified in notes (1) and (2) above.