EASA	EMERGENCY AIRWORTHINESS DIRECTIVE		
×	AD No.: 2009-0198-E Date: 04 September 2009		
	Note: This Emergency Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.		
This AD is issued in accordance with EC 1702/2003, Part 21A.3B. In accordance with EC 2042/2003 Artex I, Part N 3.301, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequent and person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD unless otherwise an ecified by the Agency [EC 2042/2003 Annex I, Part M.A.303] or agreed with the Authority of the State of Registry [EC 216/21/36, Article 1(4) exception].			
Type Approval Holder's Name :		Type/Model des maticu(s) :	
Agusta S.p.A.		AB139 and we 39 heicon ars	
TCDS Number : EASA.R.006			
Foreign AD : Not applicable			
Supersedure : This AD supersedes EASA aD 2008-01. Ide ed 13 August 2008			
ATA 53	Fuselage – Tail Boom - Inspection		
Manufacturer(s):	gustap.A.		
Applicability:	All A. 139 and AW139 helicopters, all serial numbers (S/N).		
Reason:	Reason: During the taxiing phase, the tail boom of an AW139 helicopter bent a collapsed. The root cause of this accident is still unknown and under investigation.		
S	Evidence of debonding had been previously reported on some tail boom panels of AB/AW139 helicopters. Therefore EASA AD 2008-0157 required repetitive inspections of the tail boom assembly and, in case of debonding detection, the accomplishment of corrective actions.		
	This AD, which supersedes EASA AD 2008-0157, requires as a precautionary measure to perform repetitive inspections of the tail boom panels at closer intervals and, in case of findings, the accomplishment of the relevant corrective actions.		
Effective Date:	05 September 2009		
Required action(s)	Required as indicated,	unless accomplished previously:	
and Compliance Time(s):	(1) For all AB139 and AW139 helicopters except S/N 31006, S/N 31020, S/N 31022, S/N 31042, S/N 31136, S/N 31157, S/N 31248:		
	– Within 25 of this A	5 Flight Hours (FH) or 30 days after the effective date D, whichever occurs first, and thereafter at intervals eding 50 FH, perform detailed inspections of the tail	

	panels, in accordance with the instructions of Agusta Alert Bollettino Tecnico 139-194.	
	 (2) For AB 139 and AW139 helicopters S/N 31006, S/N 31020, S/N 3 S/N 31042, S/N 31136, S/N 31157, S/N 31248: Within 5 FH after the effective date of this AD, and therear intervals not exceeding 50 FH, perform detailed inspection the tail panels, in accordance with the instructions of A Alert Bollettino Tecnico 139-193. 	
	(3) In case of signs of debonding found during the insperious equired by paragraph (1) or (2) of this AD, before next flight, tark the tobonded areas for identification, contact Agusta for instructions and follow their corrective actions.	
Ref. Publications:	Agusta Alert Bollettino Tecnico 139-193 original ssue ruted 03 September 2009;	
	Agusta Alert Bollettino Tecnico 139-194 inginal ssue, latra 03 September 2009.	
	The use of later approved revisions of these documents is acceptable for compliance with the requirements of this A	
Remarks:	1. If requested and approprice to substantiated, EASA can approve Alternative Method Companies for this AD.	
	 The safety assessment is requested not to implement the full consultation processed and an immediate publication and notification. 	
	 Enquiries regarding this As should be referred to the Airworthiness Directives, Salety Management & Research Section, Certification Directivate and Email: <u>ADs@easa.europa.eu</u>. 	
	 4 or any question concerning the technical content of the requirements in this (D, please contact Agusta S.p.A. Equal: <u>aw139.mbx@agustawestland.com</u>. 	
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