EASA AD No.: 2010-0115

## AD No.: 2010-0115 Date: 17 June 2010 Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.

This AD is issued in accordance with EC 1702/2003, Part 21A.3B. In accordance with EC 2042/2003 Annex I, Part M.A.301, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [EC 2042/2003 Annex I, Part M.A.303] or agreed with the Authority of the State of Registry [EC 216/2008, Article 14(4) exemption].

|                               | an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [EC 2042/2003 Annex I, Part M.A.303] or agreed with the Authority of the State of Registry [EC 216/2008, Article 14(4) exemption]. |  |  |
|-------------------------------|--|--|--|
| Type Approval Holder's Name : |  | Type/Model designation(s) :  |  |
| SAAB AB, Saab Aerosystems     |  | 2000 aeroplanes  |  |
| TCDS Number: EASA.A.069       |  |  |  |
| Foreign AD: Not applicable    |  |  |  |
| Supersedure: None             |  |  |  |
|                               |  |  |  |
| ATA 55                        | Stabilizers – Horizontal Stabilizer Rear Spar Upper Cap & Wiring Harness – Inspection / Modification   |  |  |
|                               |  |  |  |
| Manufacturer(s):              | SAAB AB, Saab Aerosystems (formerly Saab Aircraft AB).   |  |  |
| Applicability:                | Model 2000, all serial numbers   |  |  |
|                               | stabilizer of SAAB 2000 inboard elevator hinge w wired through the lighten   | and on the rear spar upper cap of the horizontal aeroplanes. The affected areas are adjacent to the where the electrical wiring harnesses are located and an hing holes. The upper spar cap is a primary important to the structural integrity of the horizontal |  |
| Reason:                       | Corrosion damage in these areas, if not detected and corrected, can result in a starting point for future crack propagation, which would impair the integrity of the horizontal stabilizer upper spar cap structure.   |  |  |
|                               | inspection (DVI) of the L<br>the inboard elevator hing   | e above, this AD requires a detailed visual H and RH horizontal stabilizer rear spar adjacent to ge and the harnesses installed in the adjacent voluted tubing on the harness, and corrective dings.   |  |
| Effective Date:               | 01 July 2010   |  |  |

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| Required Action(s)<br>and Compliance<br>Time(s): | Required as indicated, unless accomplished previously:  |  |
|--|---|--|
|  | (1) Within 12 months after the effective date of this AD, accomplish a DVI of the LH and RH horizontal stabilizer and the harness installed in the adjacent area, and install convoluted tubing on the harness, in accordance with the instructions of SAAB SB 2000-55-013.                                     |  |
|  | (2) If, during the DVI as required by paragraph (1) of this AD, corrosion is<br>detected, before next flight, contact SAAB (address indicated in SB) to<br>obtain approved repair instructions and accomplish those instructions<br>accordingly.  |  |
| Ref. Publications:                               | SAAB SB 2000-55-013 original issue dated 06 July 2009.  |  |
|  | The use of later approved revisions of this document is acceptable for compliance with the requirements of this AD.   |  |
| Remarks :  | <ol> <li>If requested and appropriately substantiated, EASA can approve<br/>Alternative Methods of Compliance for this AD.</li> </ol>   |  |
|  | <ol> <li>This AD was posted on 17 May 2010 as PAD 10-045 for consultation<br/>until 14 June 2010. No comments were received during the consultation<br/>period.</li> </ol>  |  |
|  | <ol> <li>Enquiries regarding this AD should be referred to the Airworthiness<br/>Directives, Safety Management &amp; Research Section, Certification<br/>Directorate, EASA. E-mail <a href="mailto:ADS@easa.europa.eu">ADS@easa.europa.eu</a></li> </ol>  |  |
|  | <ol> <li>For any question concerning the technical content of the requirements in<br/>this AD, please contact:<br/>Saab AB, Saab Aerosystems, Bröderna Ugglas gata,<br/>SE-581 88 Linköping, Sweden<br/>Telephone: +46 13 185591, Fax: +46 13 184874<br/>E-mail: technical.support@sat.saabgroup.com</li> </ol> |  |

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