EASA AD No: 2010-0194

## AD No.: 2010-0194 Date: 29 September 2010 Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.

This AD is issued in accordance with EC 1702/2003, Part 21A.3B. In accordance with EC 2042/2003 Annex I, Part M.A.301, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency IEC 2042/2003 Appex I. Part M A 3031 or agreed with the Authority of the State of Registry IEC 216/2008. Article 14(4) exemption

EC 2042/2003 Annex I, Part M.A.303] or agreed with the Authority of the State of Registry [EC 216/2008, Article 14(4) exemption].			
Type Approval Holder's Name :		Type/Model designation(s) :	
Fokker Services B.V.		F28 aeroplanes	
TCDS Number :	EASA.A.037		
Foreign AD :	Not applicable		
Supersedure :	None		
ATA 28	Fuel – Collector Tank Level Float Switch Wiring – Modification [Fuel Tank Safety]		
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Manufacturer(s):	Fokker Aircraft B.V. (and predecessor companies)		
Applicability:	F28 Mark 1000, 1000C, 2000, 3000, 3000C, 3000R, 3000RC and 4000 aeroplanes, all serial numbers.		
Reason:	Prompted by an accident of a Boeing 747-131 (flight TWA800), the Federal Aviation Administration (FAA) has published Special Federal Aviation Regulation (SFAR) 88, and the Joint Aviation Authorities (JAA) has published Interim Policy INT/POL/25/12. The review conducted by Fokker Services on the Fokker F28 Type Design in response to these regulations revealed that, under certain failure conditions, a short circuit may develop in the collector tank level float switch wiring. Such a short circuit may result in an ignition source in the tank vapour space.		
	This condition, if not consequent loss of t	t corrected, could result in a wing fuel tank explosion and the aeroplane.	
		scribed above, this AD requires the installation of a fuse ction in the collector tank level float switch wiring.	
Effective Date:	13 October 2010		

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Required Action(s) and Compliance Time(s):	Required as indicated, unless accomplished previously.	
	(1) Within 24 months after the effective date of this AD, install the fuses packed in jiffy junctions in accordance with the Accomplishment Instructions of Fokker Services Service Bulletin (SB) SBF28-28-049.	
	(2) After modification of an aeroplane as required by paragraph (1) of this AD, make certain that the fuses packed in jiffy junctions remain installed on that aeroplane, in compliance with the requirements of this AD.	
	(3) Compliance with the requirement of paragraph (2) of this AD can be demonstrated by:	
	(3.1) Revising as follows the approved aircraft maintenance programme for which the Operator or the Owner ensures the continuing airworthiness of each operated aeroplane: incorporate the CDCCL item in accordance with the information in paragraph 1.L.(1)(c) of Fokker Services SBF28-28-049,	
	and	
	(3.2) Complying with the approved aircraft maintenance programme described in paragraph (3.1) of this AD.	
Ref. Publications:	Fokker Services SBF28-28-049 dated 23 June 2010.	
	The use of later approved revisions of this document is acceptable for compliance with the requirements of this AD.	
Remarks:	<ol> <li>If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.</li> </ol>	
	<ol> <li>This AD was posted on 30 August 2010 as PAD 10-089 for consultation until 27 September 2010. No comments were received during the consultation period.</li> </ol>	
	<ol> <li>Enquiries regarding this AD should be referred to the Airworthiness Directives, Safety Management &amp; Research Section, Certification Directorate, EASA. E-mail <a href="mailto:ADS@easa.europa.eu">ADS@easa.europa.eu</a>.</li> </ol>	
	<ol> <li>For any question concerning the technical aspects of the requirements in this AD, please contact:         Fokker Services B.V., Technical Services Dept., P.O.Box 231, 2150 AE Nieuw-Vennep, The Netherlands; telephone (31) 252-627-350; facsimile (31) 252-627-211; e-mail: technicalservices.fokkerservices@fokker.com         The referenced publication can be downloaded from www.myfokkerfleet.com</li> </ol>	

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