EASA	AIRWORTHINESS DIRECTIVE	
1	AD No.: 2011-0033	
	Date: 01 March 2011	
	Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.	
continuing airworthiness of an air an aircraft to which an AD applie	craft shall be ensured by accom s, except in accordance with the	3B. In accordance with EC 2042/2003 Annex I, Part M.A.301, the plishing any applicable ADs. Consequently, no person may operate requirements of that AD unless otherwise specified by the Agency y of the State of Registry [EC 216/2008, Article 14(4) exemption].
Type Approval Hold	er's Name :	Type/Model designation(s) :
EUROCOPTER	AS350 Helicopters	
TCDS Number : EASA R.008		
Foreign AD : Not applicable		
Supersedure : None		
ATA 29	Hydraulic Power – Forward Hydraulic Servo Control Hoses / Protection Installation	
Manufacturer(s):	Eurocopter (formerly Eurocopter France, Aerospatiale)	
Applicability:	Applicability: AS 350 B, BA, BB, D, B1, B2, and B3 helicopters, all serial numbers, with single hydraulic power system and the following forward (pitch) servo-con hydraulic hoses installed:	
	- P/N 704A34-412-033	(other reference MP/N 675-102-05-01);
	- P/N 704A34-412-035	(other reference MP/N 675-102-06-01).
Reason:	helicopter. The fire was hydraulic hose, which	nain gearbox compartment occurred on an AS350B2 s caused by ignition of hydraulic fluid leaking from a had been damaged following an electrical fault in a ompartment that is not fire protected.
	when undetected or if result in loss of hydrau effects and damage to not prevented, could le structural damage and	hain gearbox compartment during a continued flight, a landing could not be performed immediately, can lics, possible shut down of the engine because of fire the Main Rotor (MR) control system. This condition, if ead to loss of MR control, potential power loss, propagation of fire into the cabin or other quently with loss of control of the helicopter and ccupants.
		bed above, this AD requires installation of protection oses as a final solution.
Effective Date:	15 March 2011	

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Required Action(s) and Compliance Time(s):	Required as indicated, unless accomplished previously:		
	 Within 30 days after the effective date of this AD install protection sleeves P/N 706A34-402-225 and 706A34-402-224 on hydraulic hoses P/N 704A34-412-033 and 704A34-412-035 respectively in accordance with the instructions of paragraph 2.B.2 of the Alert Service Bulletin (ASB) No. 29.00.13. 		
	(2) Helicopters equipped in production with modification 074238 are in compliance with the requirements of paragraph (1) of this AD.		
	(3) After the effective date of this AD, do not install on any helicopter hydraulic hoses P/N 704A34-412-033 and 704A34-412-035 without protection sleeves P/N 706A34-402-225 and 706A34-402-224 respectively.		
Ref. Publications:	EUROCOPTER AS350 B, BA, BB, D, B1, B2, B3 ASB No. 29.00.13 Revision 0, dated 26 July 2010 or later approved revisions.		
Remarks :	 If requested and appropriately substantiated, EASA can accept Alternative Methods of Compliance for this AD. 		
	 The required actions and the risk allowance have granted the issuance of a Final AD with Request for Comments, postponing the public consultation process after publication. 		
	 Enquiries regarding this AD should be referred to the Airworthiness Directives, Safety Management & Research Section, Certification Directorate, EASA. E-mail <u>ADs@easa.europa.eu</u>. 		
	 For any question concerning the technical content of the requirements in this AD, please contact: EUROCOPTER (STDI) - Aéroport de Marseille Provence 13725 Marignane Cedex - France. Tel: +33 (0) 4 42 85 97 97 - Fax: +33 (0) 4 42 85 99 66. E-mail: <u>Directive.technical-support@eurocopter.com</u> 		