


<b>EASA</b>	<b>AIRWORTHINESS DIRECTIVE</b>
	<p><b>AD No.: 2011-0052</b></p> <p><b>Date: 24 March 2011</b></p> <p>Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.</p>
<p>This AD is issued in accordance with EC 1702/2003, Part 21A.3B. In accordance with EC 2042/2003 Annex I, Part M.A.301, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of that Airworthiness Directive unless otherwise specified by the Agency [EC 2042/2003 Annex I, Part M.A.303] or agreed with the Authority of the State of Registry [EC 216/2008, Article 14(4) exemption].</p>	
<p><b>Type Approval Holder's Name:</b></p> <p>BAE Systems (Operations) Ltd</p>	<p><b>Type/Model designation(s):</b></p> <p>ATP aeroplanes</p>
TCDS Number:	EASA.A.192
Foreign AD:	Not applicable
Supersedure:	This AD supersedes EASA AD 2010-0095 dated 26 May 2010.
<b>ATA 05</b>	<b>Time Limits / Maintenance Checks – Airworthiness Limitations – Amendment / Implementation</b>
Manufacturer(s):	British Aerospace plc, British Aerospace (Commercial Aircraft) Ltd
Applicability:	ATP aeroplanes, all serial numbers.
Reason:	<p>The ATP Aircraft Maintenance Manual (AMM), includes the following Chapters:</p> <ul style="list-style-type: none"> <li>- 05-10-11 “Mandatory Life Limitations (Airframe – Systems)”,</li> <li>- 05-10-12 “Mandatory Life Limitations (Airframe – Structures)”,</li> <li>- 05-10-14 “Mandatory Life Limitations (Powerplant/Engine/APU – Systems)”,</li> <li>- 05-10-15 “Mandatory Life Limitations (Powerplant/Engine/APU – Structures)”,</li> <li>- 05-10-17 Structurally Significant Items (SSI's),</li> <li>- 05-20-00 Critical Design Configuration Control Limitations (CDCCL) – Fuel System, and,</li> <li>- 05-23-00 Certification Maintenance Requirements.</li> </ul> <p>Compliance with these chapters has been identified as mandatory actions for continued airworthiness and EASA AD 2010-0095 was issued to require operators to comply with those instructions.</p> <p>Since the issuance of EASA AD 2010-0095, BAE Systems (Operations) Ltd amended the AMM to include a revision to Chapter 05-10-17 that amends</p>

	<p>the task activity for SSI 53-10-301 to include an additional magnetic particle inspection. Failure to comply with this revision constitutes an unsafe condition.</p> <p>For the reason described above, this EASA AD retains the requirements of EASA AD 2010-0095, which is superseded, and requires the implementation of the new or more restrictive maintenance requirements and/or airworthiness limitations as specified in the defined parts of Chapter 05 of the AMM at Revision 86.</p>
Effective Date:	07 April 2011
Required Action(s) and Compliance Time(s):	<p>(1) Within 3 months after the effective date of this AD, comply with all applicable airworthiness limitations and maintenance requirements included in the following chapters of ATP AMM at Revision 86:</p> <ul style="list-style-type: none"> <li>- 05-10-11 "Mandatory Life Limitations (Airframe – Systems)",</li> <li>- 05-10-12 "Mandatory Life Limitations (Airframe – Structures)",</li> <li>- 05-10-14 "Mandatory Life Limitations (Powerplant/Engine/APU – Systems)",</li> <li>- 05-10-15 "Mandatory Life Limitations (Powerplant/Engine/APU – Structures)",</li> <li>- 05-10-17 Structurally Significant Items (SSI's),</li> <li>- 05-20-00 Critical Design Configuration Control Limitations (CDCCL) – Fuel System, and,</li> <li>- 05-23-00 Certification Maintenance Requirements.</li> </ul> <p>(2) Compliance with the requirements of paragraph (1) of this AD can be demonstrated by:</p> <p>(2.1) Revising as follows, unless accomplished previously, the approved Aircraft Maintenance Programme from which the Operator or the Owner ensures the continuing airworthiness of each operated aeroplane: Incorporate applicable maintenance requirements and associated airworthiness limitations included in the chapters of ATP AMM at Revision 86 listed in the paragraph (1) of this AD, and,</p> <p>(2.2) Complying with the approved Aircraft Maintenance Programme described in paragraph (2.1) of this AD.</p>
Ref. Publications:	<p>BAE Systems (Operations) Limited ATP AMM, Chapters 05-10-11, 05-10-12, 05-10-14, 05-10-15, 05-10-17, 05-20-00 and 05-23-00 at Revision 86, dated 15 June 2010.</p> <p>The use of later approved revisions of this document is acceptable for compliance with the requirements of this AD.</p>
Remarks :	<ol style="list-style-type: none"> <li>1. If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.</li> <li>2. This AD was posted on 28 January 2011 as PAD 11-007 for consultation until 25 February 2011. No comments were received during the consultation period.</li> <li>3. Enquiries regarding this AD should be referred to the Airworthiness Directives, Safety Management &amp; Research Section, Certification Directorate, EASA. E-mail <a href="mailto:ADs@easa.europa.eu">ADs@easa.europa.eu</a>.</li> </ol>

- |  |  |
|--|--|
|  | <p>4. For any question concerning the technical content of the requirements in this AD, please contact:<br/>BAE Systems (Operations) Ltd, Customer Information Department,<br/>Prestwick International Airport, Ayrshire, KA9 2RW, Scotland, United Kingdom; Telephone +44 1292 675207, Facsimile +44 1292 675704;<br/>E-mail: <a href="mailto:RApublications@baesystems.com">RApublications@baesystems.com</a>.</p> |
|--|--|

SUPERSEDED