EASA	AIRWORTHINESS DIRECTIVE				
	AD No.: 2011-0170				
	Date: 07 September	2011			
	Regulation (EC) No 216/2008	Directive (AD) is issued by EASA, acting in accordance with 3 on behalf of the European Community, its Member States and of 5 that participate in the activities of EASA under Article 66 of that			
This AD is issued in accordance with EC 1702/2003, Part 21A.3B. In accordance with EC 2042/2003 Annex I, Part M.A.301, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [EC 2042/2003 Annex I, Part M.A.303] or agreed with the Authority of the State of Registry [EC 216/2008, Article 14(4) exemption].					
Type Approval Holder's Name :		Type/Model designation(s) :			
AIRBUS		A330 and A340-200/-300 aeroplanes			
TCDS Numbers:	EASA.A.004, EASA.A.015				
Foreign AD:	Not applicable				
Supersedure:	None				
ATA 32	Landing Gear – Main I Life Limit Following R	Landing Gear (MLG) Wheel Axle - Reduced epair			
Manufacturer(s):	Airbus (formerly Airbus Ind	dustrie)			
Applicability:	A330-302, A330-303, A33 and A330-343 aeroplanes	02, A330-203, A330-223, A330-243, A330-301, 30-321, A330-322, A330-323, A330-341, A330-342 s, all manufacturer serial numbers, except those on 54500 has been embodied in production.			
	Airbus A340-211, A340-212, A340-213, A340-311, A340-312 and A340-313 aeroplanes, all manufacturer serial numbers, except those on which Airbus modification 54500 has been embodied in production.				
Reason:	Maintenance repair organ	ort via Airbus and Messier-Bugatti-Dowty Ltd, from a isation, concerning a specific repair, accomplished es. Investigations revealed that the axles have been a small as 0.4 mm.			
	these parts, possibly affect analyses were performed, affected MLG wheel axles	ected, has a detrimental effect on the fatigue lives of cting the structural integrity of the aeroplane. Fatigue , the results of which indicated that the life limit of the s must be reduced to below the one stated in the rworthiness Limitation Section (ALS) Part 1.			
		above, this AD requires the replacement of the eaching the new reduced demonstrated life limit.			
Effective Date:	21 September 2011				

Required Action(s)	Required as indicated, unless accomplished previously:		
and Compliance Time(s):	axle having a Part Number (P/ Appendix 1 Part 1 of Airbus All	affected MLG wheel axle is defined as a MLG N) and Serial Number (S/N) listed in the Operator Telex (AOT) A330-32A-3256, or of s applicable to the aeroplane type.	
	AD, whichever occurs late serviceable part in accord	e specified in paragraph (1.1) or (1.2) of this er, replace the affected MLG wheel axle by a lance with the instructions of Airbus AOT A330- A340-32A-4292, as applicable to the aeroplane	
	whichever occurs f	lation of the flight hours (FH) or landings (LDG), irst, as defined in Table 1 of this AD, as plane type, model and Weight Variant (WV).	
	Table 1 - Post repair MLG Wheel Axle life limit		
	Affected aeroplanes	Post-repair MLG Wheel Axle Life Limit (see Note 1)	
	A340-311/-312/-313 WV00 series	4 700 LDG or 22 250 FH	
	A340-211/-212/-213 WV00 series	4 600 LDG or 29 000 FH	
	A340-313 WV02 and WV05 series	3 950 LDG or 16 900 FH	
	A330-301/-321/-322/- 341/-342 WV00 and WV01 series	5 050 LDG or 15 200 FH	
	A330-201/-202/-203/- 223/-243 WV02, WV05 and WV06 series	4 450 LDG or 17 900 FH	
	A330-301/-302/-303/- 323/-342/-343 WV02 and WV05 series	5 150 LDG or 13 450 FH	
	Note 1: The Post repair MLG Wheel Axle life limit must be counted from the date of installation of the MLG wheel axle on an aeroplane which occurs after the date of repair specified in the Appendix 1 Part 1 of Airbus AOT A330-32A-3256 or Airbus AOT A340-32A-4292, as applicable to the aeroplane type.		
	or,		
	 (1.2) Within 24 months after the effective date of this AD, without exceeding the FH or LDG, whichever occurs first, as defined in Table 2 of this AD, as applicable to aeroplane type, model and WV. 		
		Table 2	
	Post repair MLG Wheel Axle FH or LDG not to be exceeded		
	Affected aeroplanes	Post-repair MLG Wheel Axle FH or LDG, not to be exceeded (see Note 2)	
	A340-311/-312/-313 WV00 series	7 830 LDG or 37 080 FH	
	A340-211/-212/-213 WV00 series	7 660 LDG or 48 330 FH	
	A340-313 WV02 and	6 580 LDG or 28 160 FH	

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	WV05 series
	A330-301/-321/-322/- 341/-342 WV00 and 8 410 LDG or 25 330 FH WV01 series
	A330-201/-202/-203/- 223/-243 WV02, WV05 and WV06 series 7 410 LDG or 29 830 FH
	A330-301/-302/-303/- 323/-342/-343 WV02 8 580 LDG or 22 580 FH and WV05 series
	Note 2 : The Post repair MLG Wheel Axle FH or LDG not to be exceeded must be counted from the date of installation of the MLG wheel axle on an aeroplane which occurs after the date of repair specified in the Appendix 1 Part 1 of Airbus AOT A330-32A-3256 or of Airbus AOT A340-32A-4292, as applicable to the aeroplane type.
	(2) After the effective date of this AD, an affected MLG wheel axle can be installed on an aeroplane, provided the MLG wheel axle has not exceeded the limits defined in Table 1 of this AD and it is replaced with a serviceable part before reaching the life limit defined in Table 1 of this AD.
	Note 3: After removal from an aeroplane, an affected MLG wheel axle that has reached its life limit is considered an unserviceable part.
	Note 4: The term "Life limit" used in this AD has to be understood as a post-repair life limit.
Ref. Publications:	Airbus AOT A330-32A-3256 original issue, dated 24 August 2011.
	Airbus AOT A340-32A-4292 original issue, dated 24 August 2011.
	The use of later approved revisions of these documents is acceptable for compliance with the requirements of this AD.
Remarks :	1. If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.
	 The required actions and the risk allowance have granted the issuance of a Final AD with Request for Comments, postponing the public consultation process after publication.
	3. Enquiries regarding this AD should be referred to the Safety Information Section, Executive Directorate, EASA. E-mail <u>ADs@easa.europa.eu</u> .
	 For any question concerning the technical content of the requirements in this AD, please contact: AIRBUS – Airworthiness Office – EIAL, Fax: + 33 5 61 93 45 80 or + 33 5 61 93 44 51; E-mail: <u>airworthiness.A330-A340@airbus.com</u>.
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