AD No.: 2011-0219 Date: 11 November 2011 Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.

This AD is issued in accordance with EC 1702/2003, Part 21A.3B. In accordance with EC 2042/2003 Annex I, Part M.A.301, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [EC 2042/2003 Annex I, Part M.A.303] or agreed with the Authority of the State of Registry [EC 216/2008, Article 14(4) exemption].

Type Approval Holder's Name : SAAB AB, Aeronautics		Type/Model designation(s): SF340A and 340B aeroplanes
Foreign AD :	Not applicable	
Supersedure :	None	
ATA 27	Flight Controls – Sta Replacement	ll Warning/Identification System –
Manufacturer(s):	SAAB AB, Aeronautics (formerly SAAB Aerosystems).	
Applicability:	Model SF340A aeroplanes, serial numbers (s/n) 004 through 159 inclusive, and Model 340B aeroplanes, s/n 160 through 459 inclusive, except aeroplanes that have SAAB Modification (Mod.) No. 2650 and/or Mod. No. 2859 installed. Note: This AD does not apply to aeroplanes with s/n 170, 342, 362, 363, 36 372, 379, 385, 395, 405, 409, 431 and 455.	
Reason:	372, 379, 385, 395, 405, A few natural stall events have been experienced of	409, 431 and 455. s, specifically when operating in icing conditions, on SAAB 340 series aeroplanes, without receiving condition, if not corrected, could result in loss of
Reason:	A few natural stall events have been experienced or prior stall warning. This control of the aeroplane. To address this potential	409, 431 and 455. s, specifically when operating in icing conditions, on SAAB 340 series aeroplanes, without receiving condition, if not corrected, could result in loss of
Reason:	A few natural stall events have been experienced or prior stall warning. This control of the aeroplane. To address this potential incorporating improved so SAAB have issued Serving which include instruction (SWC) with a new SWC, system includes stall war	409, 431 and 455. s, specifically when operating in icing conditions, on SAAB 340 series aeroplanes, without receiving condition, if not corrected, could result in loss of I unsafe condition, a modified stall warning system
Reason:	A few natural stall events have been experienced or prior stall warning. This control of the aeroplane. To address this potential incorporating improved so shad have issued Serving which include instruction (SWC) with a new SWC, system includes stall was conditions, which are active to the reasons described SWC, by installing new so with basic wing tip, and in	409, 431 and 455. s, specifically when operating in icing conditions, on SAAB 340 series aeroplanes, without receiving condition, if not corrected, could result in loss of unsafe condition, a modified stall warning system stall warning logic, has been developed. Ice Bulletin (SB) 340-27-098 and SB 340-27-099, is to replace the present Stall Warning Computer, and instructions to activate the new SWC. The nearning curves optimized for operation in icing

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Required Action(s)	Required as indicated, unless accomplished previously:		
and Compliance Time(s):	(1) Within 24 months after the effective date of this AD, accomplish the following actions, as applicable to aeroplane configuration:		
	(1.1) For aeroplanes with basic wing tip, replace the SWC P/N 0020AK or P/N 0020AK1, or P/N 0020AK2, or P/N 0020AK4, with a new SWC P/N 0020AK6, in accordance with the instructions of SAAB SB 340-27-098.		
	(1.2) For aeroplanes with extended wing tip, replace the SWC P/N 0020AK3 MOD 1 with a new SWC P/N 0020AK7, in accordance with the instructions of SAAB SB 340-27-099.		
	(2) Concurrent Requirement: Prior to, or concurrent with, modification of an aeroplane as required by paragraph (1) of this AD, modify the aeroplane in accordance with the instructions of SAAB SB 340-27-097 (SAAB Mod.3311).		
	(3) After modification of an aeroplane as required by paragraph (1.1) or (1.2) of this AD, as applicable, do not install any SWC P/N 0020AK, or P/N 0020AK1, or P/N 0020AK2, or P/N 0020AK4, or P/N 0020AK3 MOD 1 on that aeroplane.		
Ref. Publications:	SAAB SB 340-27-097 original issue dated 01 September 2011, or Revision 01 dated 26 September 2011, or Revision 02 dated 07 October 2011.		
	SAAB SB 340-27-098 original issue dated 01 September 2011.		
	SAAB SB 340-27-099 original issue dated 01 September 2011.		
	The use of later approved revisions of these documents is acceptable for compliance with the requirements of this AD.		
Remarks :	If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.		
	 This AD was posted on 10 October 2011 as PAD 11-106 for consultation until 07 November 2011. The Comment Response Document can be found at http://ad.easa.europa.eu/. 		
	3. Enquiries regarding this AD should be referred to the Safety Information Section, Executive Directorate, EASA. E-mail ADs@easa.europa.eu.		
	4. For any question concerning the technical content of the requirements in this AD, please contact:		
	Saab AB, Support and Services, SE-581 88 Linköping, Sweden, Fax: +46 13 184874, E-mail: saab340.techsupport@saabgroup.com .		

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