EASA AD No.: 2012-0038

AD No.: 2012-0038 Date: 12 March 2012 Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.

This AD is issued in accordance with EC 1702/2003, Part 21A.3B. In accordance with EC 2042/2003 Annex I, Part M.A.301, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [EC 2042/2003 Annex I, Part M.A.303] or agreed with the Authority of the State of Registry [EC 216/2008, Article 14(4) exemption].

Type Approval Holder's Name : SICMA AERO SEAT		Type/Model designation(s):	
		Passenger seat series 9140, 9166, 9173, 9174, 919 9188, 9196, 91B7, 91B8, 91C0, 91C2, 91C4, 91C5 91C9, 9301, 9501	
ETSO (JTSO) Auth		1O.251; DGAC France QAC 54010/SFACT/TC, 854/SFACT/TC, QAC 944322/SFACT/N,	
Foreign AD :	Not applicable		
Supersedure:	This AD supersedes DG	GAC France AD 2001-605(AB), dated 12 December 200	
ATA 25	Equipment & Furnishings – Passenger Seat Backrest Link – Inspection / Replacement		
Manufacturer(s):	Sicma Aero Seat (part of Zodiac Aerospace)		
Applicability:	Passenger seat series 9140, 9166, 9173, 9174, 9184, 9188, 9196, 91B7, 91B8, 91C0, 91C2, 91C4, 91C5, 91C9, 9301 and 9501, as identified in Annu 1 issue 3 of Sicma Aero Seat Service Bulletin (SB) 90-25-012 issue 6.		
		nger seats are known to be installed on, but no limited to and Boeing 777 aeroplanes.	
Reason:	On in-service passenger seats, some cracks were found on seat backrest lir with part number (P/N) 90-000200-104-1 and 90-000200-104-2.		
	These cracks could significantly affect the structural integrity of the seat backrests. Failures of the seat backrests could result in injury to passengers or crew members during an emergency landing.		
	To prevent this condition, a life limit was introduced on the affected backrest links and their mandatory replacement was required by DGAC France AD 2001-605(AB).		
	backrest links of sim	ssued, the seat manufacturer introduced new seat nilar design with P/N 90-000202-104-1 and P/N r passenger seat series 91B7, 91B8 and 91C5.	
	Further analysis showed that also the new seat backrest links are potentially affected by similar cracks to those identified on the backrest links with the previous design.		

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	For the reasons described above, this AD, which supersedes DGAC France AD 2001-605(AB), requires visual inspections of the seat backrest links, the accomplishment of the applicable corrective actions as well as the replacement of the backrests links before reaching their life limit.				
Effective Date:	26 March 2012				
Required Action(s) and Compliance Time(s):	Required as indicated, unless accomplished previously: (1) Inspect the seat backrest links P/N 90-000200-104-1, P/N 90-000200-104-2, P/N 90-000202-104-1 and P/N 90-000202-104-2 in accordance with the accomplishment instructions of Sicma Aero Seat SB 90-25-012 issue 6 within: (A) 6 000 seat operating hours or 2 years, whichever occurs later, from the seat manufacturing date or from the backrest link replacement, or (B) 900 seat operating hours after the effective date of this AD, but no				
	later than 5 months after the effective date of this AD,				
	whichever occurs later between (A) and (B).				
	Note: For the purpose of this AD, seat operating hours are considered to be equivalent to flight hours.				
	(2) Depending on the results of the inspection required by paragraph (1) of this AD, do the actions required by Table 1 of this AD, within the applicable compliance time(s) indicated, in accordance with the accomplishment instructions of Sicma Aero Seat SB 90-25-012 issue 6:				
	INSPECTION RESULTS	ACTION			
	not to ex hours o	the inspection at intervals acceed 900 seat operating r 5 months since the last on, whichever occurs later			
	whose length does not exceed the lock-out pin-hole (see applicable which	the cracked link(s) within rating hours or 3 months, ever occurs later, after crack identification			
	The link(s) has (have) crack(s) whose length exceeds the lock-out pin-hole (see applicable figures on Sicma Aero Seat SB 90-25-012 issue 6)	the cracked link(s) before the next flight			
	Table 1	_			
	(3) In any case after accomplishment of the requirements of paragraphs (1) and (2) of this AD, replace all seat backrest links P/N 90-000200-104-1, P/N 90-000200-104-2, P/N 90-000202-104-1 and P/N 90-000202-104-2 in accordance with the accomplishment instructions of Sicma Aero Seat SB 90-25-012 issue 6 within:				
	(C) 12 000 seat operating hours or 4 years, whichever occurs later, from the seat manufacturing date or from the backrest link replacement,				

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	or	
	(D) 3 500 seat operating hours after the effective date of this AD, but no later than 18 months after the effective date of this AD,	
	whichever occurs later between (C) and (D).	
Ref. Publications:	Sicma Aero Seat Service Bulletin 90-25-012 issue 6, dated 25 January 2012.	
	The use of later approved revisions of this document is acceptable for compliance with the requirements of this AD.	
Remarks:	If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.	
	 This AD was posted on 07 February 2012 as PAD 12-009 for consultation until 06 March 2012. No comments were received during the consultation period. 	
	 Enquiries regarding this AD should be referred to the Safety Information Section, Executive Directorate, EASA. E-mail: ADs@easa.europa.eu. 	
	4. For any question concerning the technical content of the requirements in this AD, please contact: Sicma Aero Seat Z.I. la Limoise rue Robert Maréchal Sénior 36100 Issoudun – France Telephone: +33 (0)2 54 03 39 39; Fax: +33 (0)2 54 03 39 00 E-mail: guillaume.baron@zodiacaerospace.com and/or quentin.govindapoulle@zodiacaerospace.com.	

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