EASA

AIRWORTHINESS DIRECTIVE

AD No.: 2012-0109

Date: 21 June 2012

Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.

This AD is issued in accordance with EC 1702/2003, Part 21A.3B. In accordance with EC 2042/2003 Annex I, Part M.A.301, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [EC 2042/2003 Annex I, Part M.A.303] or agreed with the Authority of the State of Registry [EC 216/2008, Article 14(4) exemption].

Type Approval Holder's Name : Fokker Services B.V.		Type/Model designation(s): F27 Aeroplanes
TCDS Number :	EASA.A.036	
Foreign AD :	Not applicable	
		AD 2010-0113 dated 15 June 2010 and EASA AD 06, including its Correction dated 01 September 2006.
ATA 05	Time Limits / Mainte Implementation	nance Checks – Maintenance Requirements –
Manufacturer(s):	Fokker Aircraft B.V.	
Applicability:	F27 Mark 050, Mark 050	02 and Mark 0604 aeroplanes, all serial numbers
Reason:	Fokker Services published issue 3 of report SE-671 dated 4 January 2012, issue 11 of report SE-622 dated 1 May 2012 and issue 6 of report SE-525 dated 26 April 2012, which are all part of the Airworthiness Limitations Section (ALS) of the Instructions for Continued Airworthiness, referred to in Section 06, Appendix 1, of the Fokker 50/60 Maintenance Review Board (MRB) document. The complete ALS currently consists of:	
	- Airworthiness Limitation SE-622, issue 11.	nce Requirements (CMRs) – report SE-525, issue 6. on Items (ALIs) and Safe Life Items (SLIs) – report Design Configuration Control Limitations (CDCCLs) – 3.
	The instructions contain actions for continued air	ed in those reports have been identified as mandatory worthiness.
	2010-0113 and EASA A implementation of the in Instructions for Continue	ed above, this AD retains the requirements of EASA AD D 2006-0209, which are superseded, and requires the spections and limitations as specified in the ALS of the ed Airworthiness, referred to in Section 06, Appendix 1 B document, reports SE-525, SE-622 and SE-671 at sues.
Effective Date:	05 July 2012	

Required Action(s)	Required as indicated, unless accomplished previously.	
and Compliance Time(s):	(1) After the effective date of this AD, accomplish the following actions, as	
	specified in Fokker Services Reports SE-525 issue 6 (CMRs), SE-622 issue 11 (ALIs and SLIs) and SE-671, issue 3 (Fuel ALIs and CDCCLs):	
	(1.1) Replace each component before or upon reaching the applicable life limit, and	
	(1.2) Within the thresholds and intervals, accomplish all applicable maintenance tasks.	
	(2) In case of finding discrepancies (as defined in the ALS, reports SE-525, SE-622 and SE-671) during accomplishment of any task as required by paragraph (1) of this AD, within the applicable compliance time specified in the ALS, report SE-525, SE-622 or SE-671, as applicable, accomplish the applicable maintenance procedures for corrective actions in accordance with the approved maintenance documentation. If no compliance time is identified in the ALS, report SE-525, SE-622 or SE-671, as applicable, accomplish the applicable corrective actions before next flight. If a detected discrepancy is not identified in the ALS, report SE-525, SE-622 or SE-671, as applicable, before next flight, contact Fokker Services for approved instructions and accomplish those instructions accordingly.	
	(3) Compliance with the requirements of paragraphs (1) and (2) of this AD can be demonstrated by:	
	(3.1) Revising as follows the approved Aircraft Maintenance Programme (AMP) and standard practices on the basis of which the operator or the owner ensures the continuing airworthiness of each operated aeroplane:	
	Incorporate the limitations, tasks and associated thresholds and intervals described in reports SE-525 (issue 6), SE-622 (issue 11), and SE-671 (issue 3),	
	(3.2) Complying with the approved AMP described in paragraph (3.1) of this AD.	
Ref. Publications:	Fokker 50 Instructions for Continued Airworthiness, ALS reports:	
	SE-525 issue 6 dated 26 April 2012, SE-622 issue 11 dated 1 May 2012, and SE-671 issue 3 dated 4 January 2012.	
	The use of later approved revisions of these reports is acceptable for compliance with the requirements of this AD.	
Remarks :	If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.	
	 This AD was posted on 04 April 2012 as PAD 12-026 for consultation until 02 May 2011 and republished on 24 May 2012 as PAD 12-026R1 for additional consultation until 07 June 2012. No comments were received during the consultation period. 	
	 Enquiries regarding this AD should be referred to the Safety Information Section, Executive Directorate, EASA; E-mail ADS@easa.europa.eu. 	
	 For any question concerning the technical content of the requirements in this AD, please contact: Fokker Services B.V., Technical Services Dept., P.O. Box 1357, 2130 EL, Hoofddorp, The Netherlands; telephone +31-88-6280-350; facsimile +31-88-6280-111; e-mail: technicalservices@fokker.com. The referenced publication can be downloaded from www.myfokkerfleet.com. 	
	www.mytokketileet.com.	