EASA AD No.: 2013-0264-E

## EASA

## **EMERGENCY AIRWORTHINESS DIRECTIVE**

AD No.: 2013-0264-E

Date: 29 October 2013

Note: This Emergency Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.

This AD is issued in accordance with EU 748/2012, Part 21.A.3B. In accordance with EC 2042/2003 Annex I, Part M.A.301, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [EC 2042/2003 Annex I, Part M.A.303] or agreed with the Authority of the State of Registry [EC 216/2008, Article 14(4) exemption].

<b>Design Approva</b> XTREMEAIR Gmb	al Holder's Name: bH	Type/Model designation(s): XA41, XA42 aeroplanes
TCDS Number:	EASA.A.507	
Foreign AD:	Not applicable	
Supersedure:	None	
ATA 71	Powerplant – Engine Mount – Inspection / Replacement	
Manufacturer(s):	XtremeAir GmbH	
Applicability:	XA42, and XA41 aeroplanes, all models, all serial numbers	
Reason:	A crack was reported by an operator in a weld seam between the lower left landing gear attachment bearing and the lower engine mount to firewall attachment plate. Further investigation showed that all engine mounts from a specific supplier may be affected.	
	This condition, if not detected and corrected, could lead to engine separation of the aeroplane and could severely affect the structural integrity of the aeroplane.	
	To address this unsafe condition, XtremeAir GmbH issued Service Bulletin SB-2013-008-A.03 to provide inspection and replacement instructions for the engine mount.	
	For the reason described above, this AD requires repetitive inspection of engine mounts and, in case of findings, replacement of the engine mount.	
Effective Date:	31 October 2013	
Required Action(s) and Compliance Time(s):	(1) For aeroplanes on w 151, manufactured b after the effective dat exceed 10 FC, perfor cracks in accordance	ess accomplished previously: hich an engine mount Part Number (P/N) XA42-7120- y Szel-Tech, is installed, within 10 flight cycles (FC) te of this AD, and, thereafter, at intervals not to rm a visual inspection of all engine mount welds for with the instructions of XtremeAir SB-2013-008- pection is not sufficient to determine cracks, perform

EASA AD No.: 2013-0264-E

	a dye penetrant inspection in accordance with the instructions of XtremeAir SB-2013-008-A.03.	
	(2) If, during any inspection as required by paragraph (1) of this AD, cracks are found, before next flight, replace the engine mount with a serviceable part (P/N XA42-7120-151) in accordance with the instructions of XtremeAir SB-2013-008-A.03.	
	Note: For the purpose of this AD, a serviceable part is a part which was inspected and, depending on findings, corrected by XtremeAir.	
	(3) Unless the engine mount was replaced with a serviceable part as required by paragraph (2) of this AD, within 4 months or before next aerobatic flight, whichever occurs first after the effective date of this AD, replace the engine mount with a serviceable part in accordance with the instructions of XtremeAir SB-2013-008-A.03.	
	(4) Replacement of the engine mount as required by paragraph (2) or (3) of this AD constitutes terminating action for the requirements of this AD.	
Ref. Publications:	XtremeAir SB-2013-008 issue A.03 dated 25 October 2013.	
	The use of later approved revisions of this document is acceptable for compliance with the requirements of this AD.	
Remarks:	If requested and appropriately substantiated, EASA can approve     Alternative Methods of Compliance for this AD.	
	<ol> <li>The results of the safety assessment have indicated the need for immediate publication and notification, without the full public consultation process.</li> </ol>	
	<ol> <li>Enquiries regarding this AD should be referred to the Safety Information Section, Executive Directorate, EASA. E-mail: <a href="mailto:ADs@easa.europa.eu">ADs@easa.europa.eu</a>.</li> </ol>	
	<ol> <li>For any question concerning the technical content of the requirements in this AD, please contact:</li> </ol>	
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