EASA AD No.: 2014-0074

AD No.: 2014-0074 Date: 21 March 2014 Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.

This AD is issued in accordance with EU 748/2012, Part 21.A.3B. In accordance with EC 2042/2003 Annex I, Part M.A.301, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [EC 2042/2003 Annex I, Part M A 303] or agreed with the Authority of the State of Registry [EC 216/2008 Article 14/4) exemption.

Design Approval Holder's Name:		Type/Model designation(s):
ATR-GIE AVION RÉGIONAL	S de TRANSPORT	ATR 72 and ATR 42 aeroplanes
TCDS Number:	EASA A.084	
Foreign AD:	Not applicable	
Supersedure:	None	
ATA 32	Landing Gear – Ma Replacement	nin Landing Gear Hinge Pin – Inspection /
Manufacturer(s):		ransport Régional (formerly Aerospatiale – Aeritalia, , Aerospatiale ATR – ALENIA, EADS ATR – Alenia)
Applicability:	ATR 42 and ATR 72 aeroplanes, all certified models, all manufacturer serial numbers.	
Reason:	number (P/N) D61000 issued AD 96-131-064	rupture of Main Landing Gear (MLG) rear hinge pin part of encountered in service in 1994 and 1996, DGAC France (B) for ATR 42 aeroplanes and AD 96-096-029 (B) for require inspection and, depending on findings, corrective
	Since those ADs were issued, new occurrences of cracked rear hinge pin P/N D61000 were reported on ATR72 MLG.	
	subjected to a non-de	ent investigation revealed that the affected pins were tected thermal abuse done in production during grinding showed that other MLG pin P/N's could be affected by ity.
		letected and corrected, could lead to MLG structural ing in collapse of the MLG and consequently injury to the plane.
	For the reasons described on findings, replacement	ribed above, this AD requires inspection and, depending ent of affected pins.

EASA AD No.: 2014-0074

Effective Date:	04 April 2014	
Required Action(s) and Compliance Time(s):	Required as indicated, unless accomplished previously:	
	For ATR 72 aeroplanes:	
	(1) Within 12 months after the effective date of this AD, identify the serial number (S/N) of the left hand (LH) and right hand (RH) MLG rear hinge pin P/N D61000.	
	(2) If, during the identification as required by paragraph (1) of this AD, a MLG rear hinge pin P/N D61000 having a S/N listed in Messier Bugatti Dowty Service Bulletin (SB) 631-32-213 or SB 631-32-216 Revision 1, as applicable, is found to be installed, within 12 months after the effective date of this AD, replace the MLG rear hinge pin P/N D61000 with a serviceable part in accordance with of Messier Bugatti Dowty SB 631-32-213 or SB 631-32-216, as applicable.	
	Note 1: For the purpose of paragraph (2) of this AD, a serviceable MLG rear hinge pin is a part that:	
	- is outside of the identified batch as listed in Messier Bugatti Dowty SB 631-32-213 or SB 631-32-216 Revision 1, as applicable, or	
	 has been inspected and reconditioned in accordance with the instructions of Messier Bugatti Dowty SB 631-32-213 or SB 631-32-216, as applicable. 	
	(3) Within the compliance time, as defined in Table 1 of this AD, identify all LH and RH MLG pins having a P/N and S/N listed in Messier Bugatti Dowty SB 631-32-214 or SB 631-32-219, as applicable.	
	(4) If, during the identification as required by paragraph (3) of this AD, a MLG pin having a P/N and S/N listed in Messier Bugatti Dowty SB 631-32-214 or SB 631-32-219, as applicable, is found to be installed, within the compliance time defined in Table 1 of this AD, replace the affected MLG with a serviceable MLG, in accordance with ATR approved maintenance instructions.	
	Note 2: For the purpose of paragraph (4) of this AD, a serviceable MLG is a part incorporating pins that are:	
	- outside of the identified batch as listed in Messier Bugatti Dowty SB 631-32-214 or SB 631-32-219, as applicable, or	
	 inspected and reconditioned in accordance with the instructions of Messier Bugatti Dowty SB 631-32-214 or SB 631-32-219, as applicable. 	
	Table 1 Compliance time	
	A or B, whichever occurs first	
	A Not later than the next MLG overhaul scheduled after the effective date of this AD	
	Within 20 000 flight cycles or 9 years, whichever occurs first, accumulated since installation of MLG on an aeroplane since new or since last overhaul, as applicable	
	For ATR 42 aeroplanes	
	(5) Within the compliance time, as defined in Table 1 of this AD, identify all LH and RH MLG pins having a P/N and S/N listed in Messier Bugatti Dowty SB 631-32-215 or SB 631-32-220, as applicable.	
	(6) If, during the identification as required by paragraph (5) of this AD, a MLG pin having a P/N and S/N listed in Messier Bugatti Dowty SB 631-32-215 or SB 631-32-220, as applicable, is found to be installed, within the	

EASA AD No.: 2014-0074

	compliance time defined in Table 1 of this AD, replace the affected MLG, with a serviceable MLG in accordance with ATR approved maintenance instructions.	
	Note 3: For the purpose of paragraph (6) of this AD, a serviceable MLG is a part incorporating pins that are:	
	 outside of the identified batch as listed in Messier Bugatti Dowty SB 631-32-215 or SB 631-32-220, as applicable, or 	
	 inspected and reconditioned in accordance with the instructions of Messier Bugatti Dowty SB 631-32-215 or SB 631-32-220, as applicable. 	
	(7) A review of MLG maintenance records is acceptable to make the identification as required by paragraphs (1), (3) and (5) of this AD, provided those records can be relied upon for that purpose, and the P/N and S/N of LH and RH MLG pins can be conclusively identified from that review.	
Ref. Publications:	Messier Bugatti Dowty SB 631-32-213, dated 16 December 2013,	
	Messier Bugatti Dowty SB 631-32-214, dated 13 January 2014,	
	Messier Bugatti Dowty SB 631-32-215, dated 13 January 2014,	
	Messier Bugatti Dowty SB 631-32-216 original issue dated 30 October 2013 and Revision 1, dated 17 December 2013.	
	Messier Bugatti Dowty SB 631-32-219, dated 03 March 2014,	
	Messier Bugatti Dowty SB 631-32-220, dated 03 March 2014.	
	The use of later approved revisions of these documents is acceptable for compliance with the requirements of this AD.	
Remarks:	If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.	
	 This AD was posted on 06 February 2014 as PAD 14-034 for consultation until 06 March 2014. The Comment Response Document can be found at http://ad.easa.europa.eu. 	
	3. Enquiries regarding this AD should be referred to the Safety Information Section, Executive Directorate, EASA. E-mail: ADs@easa.europa.eu .	
	For any question concerning the technical content of the requirements in this AD, please contact: ATR - GIE Avions de Transport Régional,	
	Continued Airworthiness Service, Tel.: +33 (0)5 62 21 62 21 - Fax: +33 (0) 5 62 21 67 18; E-mail: continued.airworthiness@atr.fr.	