EASA AD No.: 2014-0141

AD No.: 2014-0141 Date: 04 June 2014 Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.

This AD is issued in accordance with EU 748/2012, Part 21.A.3B. In accordance with EC 2042/2003 Annex I, Part M.A.301, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [EC 2042/2003 Annex I, Part M.A.303] or agreed with the Authority of the State of Registry [EC 216/2008, Article 14(4) exemption].

Design Approval Holder's Name: AIRBUS		Type/Model designation(s): A318, A319 and A 20 aeroplanes
TCDS Number:	EASA.A.064	
Foreign AD:	Not applicable	
Supersedure:	None	
ATA 32	Landing Gear – Main La	n ing Gear Upper Cardan – Life Limitation
Manufacturer:	Airbus (formerly Airbus Indu	strie)
Applicability:	Airbus As 18-11, A318-112, A318-121, A318-122, A319-111, A319-112, A316-113, A319-114, A319-115, A319-131, A319-132, A319-133, A320-211, A320-212, A320-214, A320-215, A320-216, A320-231, A320-232 and A320-233 peroplanes, all manufacturer serial numbers.	
Reason:	the grease hole / main lug ir (P/N) 201163620, is listed ir	ain landing gear (MLG) fatigue test by Messier oper cardan was found with a crack, emanating from ntersection. The affected upper cardan, Part Number on the applicable Airworthiness Limitations Section trated fatigue life of 60 000 landings.
		ed, could lead to MLG upper cardan failure, possibly and subsequent damage to the aeroplane and injury
	Prompted by these findings and further to analysis, it has been decided to reduce the life limit for certain installations of the P/N 201163620 MLG upper cardan.	
		bove, this AD requires implementation of the new I replacement of any affected MLG upper cardan seded the reduced limit.
		e affected MLG upper cardan are expected to be on of the Airbus A318/A319/A320/A321 ALS Part 1.
Effective Date:	18 June 2014	

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Required Action(s)	Required as indicated, unless accomplished previously:		
and Compliance Time(s):	(1) Within 3 months after the effective date of this AD, or before exceeding the life limit as defined in Appendix 1 of this AD, as applicable, whichever occurs later, replace each MLG upper cardan P/N 201163620 with a serviceable part.		
	Note 1: For the purpose of this AD, a serviceable MLG upper cardan is a part having P/N 201163620 that has not exceeded the life limit defined in Appendix 1 of this AD, or a MLG upper cardan having a different P/N that has not exceeded the life limit defined in Airbus A318/A319/A320/A321 ALS Part 1 Revision 02.		
	(2) From the effective date of this AD, it is allowed to install a MLG upper cardan P/N 201163620 on an aeroplane, provided the part life has not exceeded the life limit defined in Appendix 1 of this AD, as applicable, and that, following installation, it is replaced before exceeding the applicable life limit as defined in Appendix 1 of this AD (see Notes 4 and 6) with a serviceable part (see Note 1 above).		
Ref. Publications:	Airbus A318/A319/A320/A321 ALS Part 1 Revision 12, dated 13 May 2011.		
	The use of later approved revisions of this document is acceptable for compliance with the requirements of this AF.		
Remarks:	If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.		
	 This AD was posted on 25 March 1014 as PAD 14-057 for consultation until 08 April 2014. The Control of Response Document can be found at http://ad.easa.europe.a.eu/. 		
	3. Enquiries regarding this A) should be referred to the Safety Information Section, Executive Directorate, EASA. E-mail: ADS@easa.europa.eu .		
	4. For any question conserning the technical content of the requirements in this AD clease contact: AIRBUS – Airworthiness Office – EIAS, Fax +2 3 5 6 93 44 51, E-mail: account.airworth-eas@airbus.com .		

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Appendix 1 – MLG Upper Cardan P/N 201163620 Life Limit

- Note 2: The life limit as specified in this Appendix (see table below) is counted in landings, accumulated by the MLG unit since its first installation on an aeroplane.
- Note 3: The life limit applies only where the installation is valid for the given aeroplane model, configuration, and weight variant (WV).
- Note 4: In case of transfer of a part between aeroplanes having a different life limitation, refer to the Airbus A318/A319/A320/A321 ALS Part 1 Revision 02 sub-part 1-0 paragraph 6 for the method to adjust the life limitation.
- Note 5: A MLG unit on which Airbus modification (mod) 26644 is embodied (i.e. post-mod) is also known as "Enhanced" landing gear. Pre-mod 26644 MLG units are identified as P/N 201375xxx Leg and Dressing Series and post-mod 26644 MLG units are identified as P/N 201582xxx Leg and Dressing series.
- Note 6: For aeroplanes and configurations not defined in the table below, the life limitation is unchanged and those as specified in Airbus A318/A319/A320/A321 ALS Part 1 at Revision 02 still apply.
- Note 7: Certain life limits were stated incorrectly in PAD 14-057, as 50 950 C instead of 50 590 FC. The Table below shows the correct values.

Affected aeroplanes	Life Limit (see Note 2 above)
A319 pre-mod 26644, excluding Corporate Jets post-mod 28238, 2 162 and 28342	50 590
A319 post-mod 26644, excluding Corporate Jets post-mod 28388, 28162 and 28342	56 480
A320 Twin pre-mod 26644, WV012, WV011, WV112 and WV018	50 590
A320 Twin post-mod 26644, WV012, WV011, WV016 and WV018	56 480
A320 Twin post-mod 26644, WV015 and WV014	42 140