EASA AD No.: 2014-0273

EASA

AIRWORTHINESS DIRECTIVE

AD No.: 2014-0273

Date: 17 December 2014

Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.

This AD is issued in accordance with EU 748/2012, Part 21.A.3B. In accordance with EC 2042/2003 Annex I, Part M.A.301, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [EC 2042/2003 Annex I, Part M.A.303] or agreed with the Authority of the State of Registry [EC 216/2008, Article 14(4) exemption].

Design Approval AIRBUS	Holder's Name:	Type/Model designation(s): A330 aeroplanes	
TCDS Number:	EASA.A.004		
Foreign AD:	Not applicable		
Supersedure:	None		
ATA 24	Electrical Power - Alternating Current Emergency Generation - Operational Procedure		
Manufacturer(s):	Airbus (formerly Airbus Indu	ustrie)	
Applicability:	Airbus A330-201, A330-202, A330-203, A330-223, A330-223F, A330-243, A330-243F, A330-301, A330-302, A330-303, A330-321, A330-322, A330-323, A330-341, A330-342 and A330-343 aeroplanes, all manufacturer serial numbers.		
Reason: The Constant Speed Motor/Generator (CSM/G), as installed of aeroplanes, is qualified for an overload condition of 9.5kVA for duration is sufficient to perform safe landing and a GO-AROU electrical load analysis revealed that the hydraulic power might to supply the CSM/G during slat/flap extension when only one running.		an overload condition of 9.5kVA for 30 minutes. This orm safe landing and a GO-AROUND. However, aled that the hydraulic power might not be sufficient	
	This condition, if not corrected, and in conjunction with the loss of main electrical system, could lead to the scenario where the crew is not clearly warned that the electrical system has switched on the battery and thus has a limited duration that would allow a safe landing.		
	Manual (AFM) Temporary Relectrical emergency configured to deploy the Landing Recovery to ON po	resafe condition, Airbus issued an Aircraft Flight Revision (TR) on A330 aeroplane to update the uration "ELEC EMER CONFIG" procedure to be ram air turbine manually before setting the position to provide sufficient hydraulic power and ler worst-case operational conditions.	
	For the reasons described a	above, this AD requires amendment of the AFM by	

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	incorporating the applicable Airbus TR.		
Effective Date:	24 December 2014		
Required Action(s) and Compliance Time(s):	Required as indicated, unless accomplished previously: (1) Within 15 days after the effective date of this AD, amend the emergency procedures section of Airbus A330 AFM to incorporate the updated "ELEC - EMER CONFIG" procedure by inserting the AFM TR as defined in Table 1 of this AD, depending on aeroplane configuration. Table 1		
	Aeroplane configuration	AFM TR	
	A330 Pre-mod 47930	TR 427 issue 1	
	A330 Post-mod 47930	TR 428 issue 1	
	Amending the applicable AFM to incorporate a later revision which includes the AFM TR as required by this AD is acceptable to comply with the requirements of paragraph (1) of this AD. (2) Concurrent with the AFM amendment as required by paragraph (1) of this AD, inform all flight crews and, thereafter, operate the aeroplane accordingly.		
Ref. Publications:	Airbus A330 AFM TR 427 Issue 1 EASA approved on 14 October 2014.		
	Airbus A330 AFM TR 428 Issue 1 EASA approved on 14 October 2014.		
	The use of later approved revisions of these documents is acceptable for compliance with the requirements of this AD.		
Remarks:	If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.		
	Based on the required actions and the compliance time, EASA have decided to issue a Final AD with Request for Comments, postponing the public consultation process until after publication.		
	3. Enquiries regarding this AD should be referred to the Safety Information Section, Certification Directorate, EASA. E-mail: ADs@easa.europa.eu .		
	4. For any question concerning the technical content of the requirements in this AD, please contact: AIRBUS SAS – Airworthiness Office – EAL. E-mail: airworthiness.A330-A340@airbus.com .		