


<b>EASA</b>	<b>PROPOSED AIRWORTHINESS DIRECTIVE</b>	
	<p><b>PAD No.: 06 – 072R1</b></p> <p><b>Date: 11 May 2006</b></p>	
No person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of that Airworthiness Directive unless otherwise agreed with the Authority of the State of Registry.		
<b>Type Approval Holder's Name :</b>		<b>Type/Model designation(s) :</b>
AIRBUS		A318, A319, A320 and A321 aircraft
TCDS Number : EASA.A.064		
Foreign AD : None		
Supersedure: F-2004-018 (EASA approval No. 2004-752)		
<b>ATA 05</b>	<b>Airworthiness Limitation Items (ALI) – Structural maintenance</b>	
Manufacturer(s):	AIRBUS, (formerly AIRBUS INDUSTRIE)	
Applicability:	<ul style="list-style-type: none"> <li>- AIRBUS A319, A320 and A321 aircraft all certified models, all serial numbers, except A319 aircraft that have received AIRBUS modifications 28238, 28162 and 28342 in production</li> </ul> <p>and</p> <ul style="list-style-type: none"> <li>- AIRBUS A318-111 and A318-112 aircraft, all serial numbers.</li> </ul>	
Reason:	<p>The certification requirements for damage-tolerant structure were provided in AIRBUS A318/A319/A320/A321 Maintenance Planning Document (MPD) subsection 9-2 (Airworthiness Limitations Items - ALI).</p> <p>The content of the MPD section 9 has been transferred into the ALS (Airworthiness Section) and the ALI are now listed in the ALS Part 2 that is approved by the European Aviation Safety Agency (EASA) - ALS required by JAR 25.1529, appendix H §25.4).</p> <p>This Airworthiness Directive (AD) mandates the ALI requirements as defined in AIRBUS Airworthiness Limitation Items document A1/SE-M4/95A.0252/96 at issue 7.</p> <p>AIRBUS document A1/SE-M4/95A.0252/96 that has been approved at issue 7 on February 7th, 2006, reduces threshold and/or interval of some existing ALI and introduces some new ALI.</p>	

	This PAD 06-072R1 introduces a limited exemption of 2 weeks to comply with the requirements of Airbus ALS Part 2 as defined in AIRBUS Airworthiness Limitation Items document AI/SE-M4/95A.0252/96 Issue 7.
Effective Date:	Proposed 31 May 2006
Compliance:	<ol style="list-style-type: none"> <li>1. Within 2 weeks from the effective date of this AD, it is mandatory to strictly adhere to the requirements of the <i>AIRBUS</i> ALS Part 2 as defined in AIRBUS Airworthiness Limitation Items document AI/SE-M4/95A.0252/96 at either: <ul style="list-style-type: none"> <li>• issue 6 approved on July 15, 2003</li> </ul> or <ul style="list-style-type: none"> <li>• issue 7 approved on February 7, 2006</li> </ul> </li> <li>2. After 2 weeks from the AD effective date, it is mandatory to strictly adhere to the requirements defined in the A318/A319/A320/A321 Airworthiness Limitation Items document AI/SE-M4/95A.0252/96 at issue 7 approved on February 7, 2006.</li> </ol>
Ref. Publications:	AIRBUS AI/SE-M4/95A.0252/96 issue 6 approved on July 15, 2003 by EASA.  AIRBUS AI/SE-M4/95A.0252/96 issue 7 approved on February 7th, 2006 by EASA.  or later approved revisions.
Remarks :	<ol style="list-style-type: none"> <li>1. If requested and appropriately substantiated the responsible EASA manager for the related product has the authority to accept Alternative Method of Compliance (AMOCs) for this AD.</li> <li>2. The closing date for comments is 19 May 2006.</li> <li>3. Enquiries regarding this AD should be addressed to Mr. M. Capaccio, AD Focal Point, Certification Directorate, EASA. E-mail <a href="mailto:ADs@easa.eu.int">ADs@easa.eu.int</a>.</li> <li>4. For any question concerning the technical content of the requirements in this AD, please contact AIRBUS - Fax 33 5 61 93 44 51</li> </ol>