


<b>EASA</b>	<b>PROPOSED AIRWORTHINESS DIRECTIVE</b>	
	<p><b>PAD No.: 06 - 072</b></p> <p><b>Date: 22 March 2006</b></p>	
No person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of that Airworthiness Directive unless otherwise agreed with the Authority of the State of Registry.		
<b>Type Approval Holder's Name :</b>		<b>Type/Model designation(s) :</b>
AIRBUS		A318, A319, A320 and A321 aircraft
TCDS Number : EASA.A.064		
Foreign AD : None		
Supersedure: F-2004-018		
<b>ATA 05</b>	<b>Airworthiness Limitation Items (ALI) – Structural maintenance</b>	
Manufacturer(s):	AIRBUS, AIRBUS INDUSTRIE	
Applicability:	<ul style="list-style-type: none"> <li>- AIRBUS A319, A320 and A321 aircraft all certified models, all serial numbers, except A319 aircraft that have received AIRBUS modifications 28238, 28162 and 28342 in production</li> </ul> <p>and</p> <ul style="list-style-type: none"> <li>- AIRBUS A318-111 and A318-112 aircraft, all serial numbers.</li> </ul>	
Reason:	<p>The certification requirements for damage-tolerant structure are provided in AIRBUS A318/A319/A320/A321 Maintenance Planning Document (MPD) subsection 9-2 (Airworthiness Limitations Items - ALI).</p> <p>The MPD subsection 9-2 is approved by the European Aviation Safety Agency (EASA) and is included in the section 9 (Airworthiness Limitation Section - ALS required by JAR 25.1529, appendix H §25.4).</p> <p>This Airworthiness Directive (AD) mandates the ALI requirements as defined in AIRBUS document AI/SE-M4/95A.0252/96 at issue 7 that is referenced in the sub-section MPD 9-2.</p> <p>AIRBUS document AI/SE-M4/95A.0252/96 that has been approved at issue 7 on February 7th, 2006, reduces threshold and/or interval of some existing ALI and introduces some new ALI.</p>	
Effective Date:	Proposed 18 May 2006	

Compliance:	<p>From the effective date of this AD, it is mandatory to strictly adhere to the requirements defined in the A318/A319/A320/A321 Airworthiness Limitation Items referenced AI/SE-M4/95A.0252/96 at issue 7 approved on February 7th, 2006.</p> <p>The corresponding documentation will have to be updated within one (1) month following the effective date of this AD.</p> <p><b>Note:</b> Tolerances for compliance with the new or revised requirements, introduced at issue 7 of the AI/SE-M4/95A.0252/96 document, are defined in this document.</p>
Ref. Publications:	<p>AIRBUS AI/SE-M4/95A.0252/96 issue 7 approved on February 7th, 2006 by EASA.</p> <p>or later approved revisions.</p>
Remarks :	<ol style="list-style-type: none"> <li>1. If requested and appropriately substantiated the responsible EASA manager for the related product has the authority to accept Alternative Method of Compliance (AMOCs) for this AD.</li> <li>2. The closing date for comments is 21 April 2006.</li> <li>3. Enquiries regarding this AD should be addressed to Mr. M. Capaccio, AD Focal Point, Certification Directorate, EASA. E-mail <a href="mailto:ADs@easa.eu.int">ADs@easa.eu.int</a>.</li> <li>4. For any question concerning the technical content of the requirements in this AD, please contact AIRBUS - Fax 33 5 61 93 44 51</li> </ol>