


<b>EASA</b>	<b>NOTIFICATION OF A PROPOSAL TO ISSUE AN AIRWORTHINESS DIRECTIVE</b>
	<p><b>PAD No.: 09-016</b></p> <p><b>Date: 05 August 2009</b></p> <p>Note: This Proposed Airworthiness Directive (PAD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.</p>
<p>In accordance with the EASA Continuing Airworthiness Procedures, the Executive Director is proposing the issuance of an EASA Airworthiness Directive (AD), applicable to the aeronautical product(s) identified below. All interested persons may send their comments, referencing the PAD Number above, to the e-mail address specified in the 'Remarks' section, prior to the consultation closing date indicated.</p>	
<p><b>Type Approval Holder's Name :</b></p> <p>AIRBUS</p>	<p><b>Type/Model designation(s) :</b></p> <p>A300-600 aeroplanes</p>
<p>TCDS Number : France No. 145</p>	
<p>Foreign AD : Not applicable</p>	
<p>Supersedure : None</p>	
<b>ATA 53</b>	<b>Fuselage – Frame 58 Butt Joint – Reinforcement</b>
Manufacturer(s):	Airbus (formerly Airbus Industrie)
Applicability:	<p>A300B4-601, A300B4-603, A300B4-605R, A300B4-620, A300B4-622, A300B4-622R, A300C4-605R variant F, A300C4-620 and A300F4-605R pre-mod 12699 aeroplanes, all serial numbers.</p> <p><b>Note:</b> For A300F4-605R post-mod 12699 and A300F4-622R aeroplane models, whose Limit of Validity is 30 000 Flight Cycles (FC), the requirements of this AD do not apply.</p>
Reason:	<p>A recent Wide spread Fatigue Damage (WFD) calculation on A300-600 aeroplanes has shown that a reinforcement of the upper fuselage circumferential joint at FR 58 is necessary to enable the aeroplane to reach the Extended Service Goal (ESG).</p> <p>The failure of the circumferential joint of the upper fuselage could affect the structural integrity of the aeroplane.</p> <p>For the reasons described above, this AD requires the reinforcement of the affected fuselage frame butt joint.</p>
Effective Date:	[TBD: 14 days after final AD issue date]

Required Action(s) and Compliance Time(s):	<p>Required as indicated, unless already accomplished:</p> <p>Before accumulation of 42 500 FC, or within 2 000 FC after the effective date of this AD, whichever occurs later, reinforce the fuselage butt joint at FR 58 in accordance with the accomplishment instructions of Airbus Service Bulletin (SB) A300-53-6146 Revision 01.</p>
Ref. Publications:	<p>Airbus SB A300-53-6146 Revision 01.</p> <p>The use of later approved revisions of this document is acceptable for compliance with requirements of this AD.</p>
Remarks :	<ol style="list-style-type: none"> <li>1. This Proposed AD will be closed for consultation on 02 September 2009.</li> <li>2. Enquiries regarding this PAD should be referred to the Airworthiness Directives, Safety Management &amp; Research Section, Certification Directorate, EASA. E-mail <a href="mailto:ADs@easa.europa.eu">ADs@easa.europa.eu</a>.</li> <li>3. For any question concerning the technical content of the requirements in this PAD, please contact: AIRBUS SAS – EAW Airworthiness Office, telephone: + 33 5 61 93 36 96, facsimile: + 33 5 61 93 44 51.</li> </ol>