


<b>EASA</b>	<b>AIRWORTHINESS DIRECTIVE</b>
	<b>AD No.: 2009-0053</b>  <b>Date: 05 March 2009</b>  Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.
This AD is issued in accordance with EC 1702/2003, Part 21A.3B. In accordance with EC 2042/2003 Annex I, Part M.A.301, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [EC 2042/2003 Annex I, Part M.A.303] or agreed with the Authority of the State of Registry [EC 216/2008, Article 24(4) exemption].	
<b>Type Approval Holder's Name :</b>	<b>Type/Model designation(s) :</b>
BAE Systems (Operations) Ltd	ATP aeroplanes
TCDS Number : United Kingdom BA23	
Foreign AD : Not applicable	
Supersedure : This AD supersedes EASA AD 2008-0141 dated 28 July 2008	
<b>ATA 05</b>	<b>Time Limits / Maintenance Checks – Airworthiness Limitations – Implementation</b>
<b>Manufacturer(s):</b> British Aerospace plc, British Aerospace (Commercial Aircraft) Ltd	
<b>Applicability:</b> ATP aeroplanes, all serial numbers.	
<b>Reasons:</b>	<p>The BAE Systems (Operations) Limited ATP Aircraft Maintenance Manual (AMM), includes Chapters 05-10-11 "Mandatory Life Limitations (Airframe – Systems)", 05-10-12 "Mandatory Life Limitations (Airframe – Structures)", 05-10-14 "Mandatory Life Limitations (Powerplant/Engine/APU – Systems)", 05-10-15 "Mandatory Life Limitations (Powerplant/Engine/APU – Structures)", 05-10-17 "Structurally Significant Items (SSI's)", 05-20-00 "Critical Design Configuration Control Limitations (CDCCL) – Fuel System" and 05-23-00 "Certification Maintenance Requirements" which have been identified as mandatory actions for continued airworthiness.</p> <p>BAE Systems (Operations) Limited has amended the AMM to include revisions to Chapters 05-10-12 and 05-10-17 that introduce some revised and corrected tasks, including one with a more restrictive limitation.</p> <p>For the reasons described above, this EASA AD retains the requirements of EASA AD 2008-0141, which is superseded, and requires the implementation of the inspections and corrective measures as specified in Chapters 05-10-11, 05-10-12, 05-10-14, 05-10-15, 05-10-17, 05-20-00 and 05-23-00 of the ATP AMM at Revision 82.</p>
<b>Effective Date:</b>	19 March 2009

<p>Required Action(s) and Compliance Time(s):</p>	<p>Required as indicated, unless accomplished previously:</p> <p>(1) Within the next 3 months after the effective date of this AD, amend the approved aircraft maintenance programme to incorporate the instructions of ATP AMM at Revision 82 Chapters 05-10-11 "Mandatory Life Limitations (Airframe – Systems)", 05-10-12 "Mandatory Life Limitations (Airframe – Structures)", 05-10-14 "Mandatory Life Limitations (Powerplant/Engine/APU – Systems)", 05-10-15 "Mandatory Life Limitations (Powerplant/Engine/APU – Structures)", 05-10-17 "Structurally Significant Items (SSI's)", 05-20-00 "Critical Design Configuration Control Limitations (CDCCL) – Fuel System" and 05-23-00 "Certification Maintenance Requirements".</p> <p>(2) Thereafter, within the thresholds and intervals indicated in this document, accomplish the tasks described in BAE Systems (Operations) Limited ATP AMM Chapters 05-10-11, 05-10-12, 05-10-14, 05-10-15, 05-10-17, 05-20-00 and 05-23-00.</p>
<p>Ref. Publications:</p>	<p>BAE Systems (Operations) Limited ATP AMM, chapters 05-10-11, 05-10-12, 05-10-14, 05-10-15, 05-10-17, 05-20-00 and 05-23-00 at Revision 82, dated 15 July 2008.</p> <p>The use of later approved revisions of this document is acceptable for compliance with the requirements of this AD.</p>
<p>Remarks :</p>	<ol style="list-style-type: none"> <li>1. If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.</li> <li>2. This AD was published on 11 February 2009 as PAD 09-028 for consultation until 04 March 2009. No comments were received during the consultation period.</li> <li>3. Enquiries regarding this AD should be referred to the Airworthiness Directives, Safety Management &amp; Research Section, Certification Directorate, EASA. E-mail <a href="mailto:ADs@easa.europa.eu">ADs@easa.europa.eu</a>.</li> <li>4. For any question concerning the technical content of the requirements in this AD, please contact: BAE Systems (Operations) Ltd, Customer Information Department, Prestwick International Airport, Ayrshire, KA9 2RW, Scotland, United Kingdom; Telephone +44 1292 675207, Facsimile +44 1292 675704; E-mail: <a href="mailto:RApublications@baesystems.com">RApublications@baesystems.com</a></li> </ol>