


EASA	NOTIFICATION OF A PROPOSAL TO ISSUE AN AIRWORTHINESS DIRECTIVE	
	PAD No.: 10 - 006 Date: 21 January 2010 Note: This Proposed Airworthiness Directive (PAD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.	
	In accordance with the EASA Continuing Airworthiness Procedures, the Executive Director is proposing the issuance of an EASA Airworthiness Directive (AD), applicable to the aeronautical product(s) identified below. All interested persons may send their comments, referencing the PAD Number above, to the e-mail address specified in the 'Remarks' section, prior to the consultation closing date indicated.	
Type Approval Holder's Name : DASSAULT AVIATION		Type/Model designation(s) : FALCON 7X
TCDS Number : EASA.A.155		
Foreign AD : Not Applicable		
Supersedure : None		
ATA 24 & 29	Electrical Power & Hydraulic Power – Electrical Wirings & Hydraulic Lines Routing – Modification(s)	
Manufacturer(s):	DASSAULT AVIATION	
Applicability:	All Falcon 7X aeroplanes, all serial numbers, except those on which modifications M964, M897, M937, M976, M1007, M1020 and M1022 have all been implemented.	
Reason:	<p>On some Falcon 7X aeroplanes, it has been determined potential low clearance between electrical wiring or hydraulic pipe and nearby structure.</p> <p>Although no in service incident has been reported, there is no certainty that the minimum clearances would be maintained over time. In the worst case, interference or contact with structure might occur and lead to electrical short circuits or fluid leakage, potentially resulting in loss of several functions essential for safe flight.</p> <p>Dassault Aviation has developed two Service Bulletins (SB) that provide corrective actions to ensure the minimum required clearance, as well as adequate protection between hydraulic pipe (SB n°092) and electrical wiring (SB n°006) and the aeroplane structure.</p> <p>This AD requires the implementation of both SBs on the affected aeroplanes.</p>	
Effective Date:	[TBD: 14 days after final AD issue date]	
Required action(s) and Compliance Time(s):	Required as indicated unless previously accomplished:	

	<p>(1) Within 10 months or 650 flight hours (FH) after the effective date of this AD, whichever occurs first, inspect the concerned wirings for condition and modify the aeroplane, in accordance with the instructions of Dassault Aviation SB F7X n° 006 ;</p> <p>(2) Within 10 months or 650 FH after the effective date of this AD, whichever occurs first, inspect the rear fuel tank panel for condition and modify the aeroplane, in accordance with the instructions of Dassault Aviation SB F7X n° 092.</p>
Ref. Publications:	<p>Dassault Aviation Service Bulletin SB F7X n° 006, initial issue ;</p> <p>Dassault Aviation Service Bulletin SB F7X n° 092, initial issue.</p> <p>The use of later approved revisions of these documents is acceptable for compliance with the requirements of this AD.</p>
Remarks:	<p>1. This Proposed AD will be closed for consultation on 18 February 2010.</p> <p>2. Enquiries regarding this PAD should be referred to the Airworthiness Directives, Safety Management & Research Section, Certification Directorate, EASA. E-mail: ADs@easa.europa.eu.</p> <p>3. For any questions concerning the technical content of the requirements in this PAD, please contact your Dassault Falcon Technical Assistance:</p> <ul style="list-style-type: none"> • For Europe, Middle East and Africa based operators: Hot Line: (33) 1 47 11 37 37 / Fax: (33) 1 47 11 89 49, • For USA, Canada and Mexico based operators: Help Desk: (1) 800-2FALCON (2325266) / Fax: (1) 201 541 4740, • All other areas: Help Desk: (1) 201 541 4747 / Fax: (1) 201 541 4740.