

EASA	AIRWORTHINESS DIRECTIVE
	AD No.: 2010-0245 Date: 26 November 2010 Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.
This AD is issued in accordance with EC 1702/2003, Part 21A.3B. In accordance with EC 2042/2003 Annex I, Part M.A.301, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [EC 2042/2003 Annex I, Part M.A.303] or agreed with the Authority of the State of Registry [EC 216/2008, Article 14(4) exemption].	
Type Approval Holder's Name : AIRBUS	Type/Model designation(s) : A340-200/-300 aeroplanes
TCDS Number : EASA.A.015	
Foreign AD : Not applicable	
Supersedure : None	
ATA 53	Fuselage – Fuselage Internal Structure at Frame 39.1 - Inspection
Manufacturer(s):	Airbus (formerly Airbus Industrie)
Applicability:	Airbus A340 aeroplanes, models -211, -212, -311 and -312 , having manufacturer serial numbers (MSN): 0002, 0003, from 0005 up to 0009 inclusive, 0011, 0013, 0014, 0015, from 0018 up to 0023 inclusive, 0025, 0026 and 0027.
Reason:	<p>Airworthiness Limitation Item (ALI) task 533105-01-01 is applicable to aeroplanes on which Airbus modification 40391 has not been embodied in production. The requirements associated to this task are applicable to aeroplanes on which Modification Proposal (MP-S10437) has not been embodied.</p> <p>Following a query from an operator, investigations revealed that some MSN, for which Airbus modification 40391 was indicated as fully embodied inside the Aircraft Inspection Report (AIR), did not have Modification Proposal (MP-S10437) which is part of this modification embodied in production.</p> <p>As a result, ALI task 533105-01-01 has not been taken into account for some MSN listed in the applicability section of this AD, which constitutes an unsafe condition.</p> <p>For the reasons described above, this AD requires repetitive special detailed inspections corresponding to ALI task 533105-01-01 and the accomplishment of the associated corrective actions, for all aeroplanes to which this task is applicable.</p>

	Airworthiness Limitation Item (ALI) task 533105-01-01 will be deleted in the next ALS Part 2 Revision.
Effective Date:	10 December 2010
Required action(s) and Compliance Time(s):	<p>Required as indicated, unless already accomplished:</p> <ol style="list-style-type: none"> (1) Before the accumulation of 13 600 total Flight Cycles (FC) or 92 100 total Flight Hours (FH), whichever occurs first, since the aeroplane first flight, and thereafter, at intervals not to exceed 7 850 FC or 53 300 FH from the last inspection, whichever occurs first, perform a special detailed inspection of both Left Hand and Right Hand Fuselage Frame 39.1 at the fastener hole area just above stringer 28, in accordance with the instructions of Airbus Service Bulletin A340-53-4184. (2) In case of any crack finding, before next flight, contact Airbus to obtain the necessary approved instructions for corrective actions and accomplish those instructions accordingly. (3) Aeroplanes which have performed ALI task 533105-01-01, are compliant with the initial inspection requirements of paragraph (1) of this AD. The repetitive inspection requirements of paragraph (1) of this AD remain applicable.
Ref. Publications:	<p>Airbus Service Bulletin A340-53-4184 at original issue.</p> <p>The use of later approved revisions of this document is acceptable for compliance with the requirements of this AD.</p>
Remarks:	<ol style="list-style-type: none"> 1. If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD. 2. This AD was posted on 08 October 2010 as PAD 10-108 for consultation until 05 November 2010. No comments were received during the consultation period. 3. Enquiries regarding this AD should be referred to the Airworthiness Directives, Safety Management & Research Section, Certification Directorate, EASA. E-mail: ADs@easa.europa.eu. 4. For any question concerning the technical content of the requirements in this AD, please contact: Airbus – Airworthiness Office – EAL; E-mail: airworthiness.A330-A340@airbus.com.