


EASA	NOTIFICATION OF A PROPOSAL TO ISSUE AN AIRWORTHINESS DIRECTIVE
	<p>PAD No.: 12-031</p> <p>Date: 19 April 2012</p> <p>Note: This Proposed Airworthiness Directive (PAD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.</p>
<p>In accordance with the EASA Continuing Airworthiness Procedures, the Executive Director is proposing the issuance of an EASA Airworthiness Directive (AD), applicable to the aeronautical product(s) identified below. All interested persons may send their comments, referencing the PAD Number above, to the e-mail address specified in the 'Remarks' section, prior to the consultation closing date indicated.</p>	
<p>Type Approval Holder's Name :</p> <p>Pilatus Aircraft Ltd.</p>	<p>Type/Model designation(s) :</p> <p>PC-12 aeroplanes</p>
TCDS Number:	EASA.A.089
Foreign AD:	Not applicable
Supersedure:	None
ATA 05	Time Limits / Maintenance Checks – Airworthiness Limitation Section – Amendment / Implementation
Manufacturer(s):	Pilatus Aircraft Ltd.
Applicability:	PC-12, PC-12/45, PC-12/47, and PC-12/47E aeroplanes, all manufacturer serial numbers.
Reason:	<p>The mandatory instructions and airworthiness limitations applicable to the Structure and Components of the PC-12 are specified in the Aircraft Maintenance Manual (AMM) under Chapter 4.</p> <p>These documents include the maintenance instructions and/or airworthiness limitations developed by Pilatus Aircraft Ltd. and approved by EASA. Failure to comply with these instructions and limitations could potentially lead to unsafe condition.</p> <p>For the reasons described above, this AD requires the implementation of more restrictive maintenance instructions and/or airworthiness limitations.</p>
Effective Date:	[TBD: 14 days after final AD issue date]
Required Action(s) and Compliance Time(s):	<p>Required as indicated unless previously accomplished:</p> <p>(1) After the effective date of this AD, accomplish the following actions, as detailed in Chapter 4 of the AMM report specified in Table 1 of this AD, as applicable:</p> <p>(1.1) Replace each component before or upon reaching the applicable life limit, and</p>

	<p>(1.2) Within the thresholds and intervals, accomplish all applicable maintenance tasks.</p> <p>(2) In case of finding discrepancies (as defined in Chapter 4 of the AMM report specified in Table 1 of this AD, as applicable) during accomplishment of any task as required by paragraph (§) (1) of this AD, within the applicable compliance time specified in Chapter 4 of the AMM report specified in Table 1 of this AD, as applicable, accomplish the applicable maintenance procedures for corrective actions in accordance with the approved maintenance documentation. If no compliance time is identified in Chapter 4 of the AMM report specified in Table 1 of this AD, as applicable, accomplish the applicable corrective actions before next flight. If a detected discrepancy is not identified in Chapter 4 of the AMM report specified in Table 1 of this AD, as applicable, before next flight, contact Pilatus Aircraft Ltd for approved instructions and accomplish those instructions accordingly.</p> <p style="text-align: center;">Table 1</p> <table border="1" data-bbox="582 660 1401 913"> <thead> <tr> <th>Aeroplane Model</th><th>AMM reference</th></tr> </thead> <tbody> <tr> <td>PC-12, PC-12/45, PC-12/47</td><td>Chapter 4 of AMM report 02049, issue 25, dated 25 January 2012</td></tr> <tr> <td>PC-12/47E</td><td>Chapter 4 of AMM report 02300, issue 8, dated 25 January 2012</td></tr> </tbody> </table> <p>(1) Compliance with the requirements of § (1) of this AD can be demonstrated by:</p> <p>(2.1) Revising as follows, unless accomplished previously, the approved Aircraft Maintenance Programme from which the operator or the owner ensures the continuing airworthiness of each operated aeroplane:</p> <p style="padding-left: 40px;">Incorporate applicable maintenance requirements and associated airworthiness limitations specified in § (1) of this AD</p> <p style="padding-left: 40px;">and</p> <p>(2.2) Complying with the approved Aircraft Maintenance Programme described in § (2.1) of this AD.</p>	Aeroplane Model	AMM reference	PC-12, PC-12/45, PC-12/47	Chapter 4 of AMM report 02049, issue 25, dated 25 January 2012	PC-12/47E	Chapter 4 of AMM report 02300, issue 8, dated 25 January 2012
Aeroplane Model	AMM reference						
PC-12, PC-12/45, PC-12/47	Chapter 4 of AMM report 02049, issue 25, dated 25 January 2012						
PC-12/47E	Chapter 4 of AMM report 02300, issue 8, dated 25 January 2012						
Ref. Publications:	<ol style="list-style-type: none"> Pilatus PC-12 Structural, Component and Miscellaneous – Airworthiness Limitations, Chapter 4 of AMM report 02049 issue 25 dated 5 January 2012, for PC-12, PC-12/45, PC-47. Pilatus PC-12 Structural, Component and Miscellaneous – Airworthiness Limitations, Chapter 4 of AMM report 02300 issue 8 dated 25 January 2012, for PC-12/47E. <p>The use of later approved revision of these documents is acceptable for compliance with the requirements of this AD.</p>						
Remarks :	<ol style="list-style-type: none"> This Proposed AD will be closed for consultation on 17 May 2012. Enquiries regarding this PAD should be referred to the Safety Information Section, Executive Directorate, EASA. E-mail ADs@easa.europa.eu. For any question concerning the technical content of the requirements in this PAD, please contact: PILATUS AIRCRAFT LTD., Customer Support Manager, CH-6371 STANS, Switzerland. Tel.: +41 (0)41 619 62 08 Fax: +41 (0)41 619 73 11. E-mail: SupportPC12@pilatus-aircraft.com. 						