EASA	NOTIFICATION OF A PROPOSAL TO ISSUE AN AIRWORTHINESS DIRECTIVE		
	PAD No.: 12-134		
	Date: 26 October 2012		
	Note: This Proposed Airworthiness Directive (PAD) is issued by EASA, acting in accordar with Regulation (EC) No 216/2008 on behalf of the European Community, its Member Stat and of the European third countries that participate in the activities of EASA under Article 66 that Regulation.		
In accordance with the EASA Continuing Airworthiness Procedures, the Executive Director is proposing the issuance of an EASA Airworthiness Directive (AD), applicable to the aeronautical product(s) identified below. All interested persons may send their comments, referencing the PAD Number above, to the e-mail address specified in the 'Remarks' section, prior to the consultation closing date indicated.			
Design Approval H	Holder's Name:	Type/Model designation(s):	
Costruzioni Aeronautiche TECNAM srl		P92 and P2002 aeroplanes	
TCDS Number: EASA A.006, EASA A.412			
Foreign AD: Not applicable			
Supersedure: None			
ATA 55	Stabilizers – Stabilator Trim Tab Hinge Supports – Inspection		
Manufacturer(s):	Costruzioni Aeronautiche TECNAM S.r.I.		
Applicability:	P92-J and P92-JS aeroplanes, all serial numbers (S/N).		
	P2002-JF and P2002-JR ae	roplanes, all S/N.	
Reason:	During a visual inspection affecting the stabilator assembly, multiple cracks were found behind the stabilator trim tab centre hinge supports Part Number (P/N) 92-7-109-1. The technical investigations are on-going, to determine the reason that led to those cracks.		
		and corrected, could affect the structural integrity possibly resulting in stabilator failure and consequent ne.	
		ostruzioni Aeronautiche TECNAM S.r.l. issued S Edition 1 st rev 0 for accomplishing repetitive m tab hinge supports.	
	of all stabilator trim tab hinge findings, the accomplishmer	bove, this AD requires repetitive visual inspections e supports P/N 92-7-109-1 and, depending on at of applicable corrective action(s). Depending on s, further AD action may follow.	
Effective Date:	[TBD: 14 days after final AD issue date]		

Required Action(s) and Compliance Time(s):	Required as indicated, unless accomplished previously.		
	(1) Within 10 flight hours (FH) or 30 days, whichever occurs first after the effective date of this AD, and thereafter, at intervals not to exceed 100 FH, inspect all Left Hand and Right Hand stabilator trim tab hinge supports P/N 92-7-109-1 in accordance with the accomplishment instructions of Costruzioni Aeronautiche Tecnam S.r.I. SB 108–CS Edition 1 st rev 0.		
	(2) The inspections required by paragraph (1) of this AD may be accomplished by the pilot owner in accordance with the provisions of Commission Regulation (EC) No 2042/2003 Annex I (Part-M).		
	(3) If, during any inspection as required by paragraph (1) of this AD, a crack is identified, before next flight, contact Costruzioni Aeronautiche Tecnam S.r.I. for repair instructions and accomplish those instructions accordingly.		
	 (4) Accomplishment of the corrective action as required by paragraph (3) does not constitute terminating action for the repetitive inspections as required by paragraph (1) of this AD. 		
Ref. Publications:	Costruzioni Aeronautiche Tecnam SB 108–CS Edition 1 st rev 0 dated 08 October 2012. The use of later approved revisions of this document is acceptable for compliance with the requirements of this AD.		
Remarks:	1. This Proposed AD will be closed for consultation on 23 November 2012.		
	 Enquiries regarding this PAD should be referred to the Safety Information Section, Executive Directorate, EASA. E-mail: <u>ADs@easa.europa.eu</u>. 		
	 For any question concerning the technical content of the requirements in this PAD, please contact: Costruzioni Aeronautiche TECNAM, Airworthiness Office, Telephone: +39 0823 620134; Fax: +39 0823 622899 E-mail: <u>airworthiness@tecnam.com</u>. 		