

<b>EASA</b>	<b>NOTIFICATION OF A PROPOSAL TO ISSUE AN AIRWORTHINESS DIRECTIVE</b>	
	<b>PAD No.: 13-006</b>  <b>Date: 08 January 2013</b>  Note: This Proposed Airworthiness Directive (PAD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.	
	In accordance with the EASA Continuing Airworthiness Procedures, the Executive Director is proposing the issuance of an EASA Airworthiness Directive (AD), applicable to the aeronautical product(s) identified below. All interested persons may send their comments, referencing the PAD Number above, to the e-mail address specified in the 'Remarks' section, prior to the consultation closing date indicated.	
<b>Design Approval Holder's Name:</b>  FOKKER SERVICES B.V.		<b>Type/Model designation(s):</b>  F27 aeroplanes
TCDS Number:      EASA.A.036		
Foreign AD:      Not applicable		
Supersedure:      None		
<b>ATA 28</b>	<b>Fuel – Fuel Tanks – Modification [Fuel Tank Safety]</b>	
Manufacturer(s):	Fokker Aircraft B.V.	
Applicability:	F27 Mark 050, Mark 0502 and Mark 0604 aeroplanes, all serial numbers.	
Reason:	<p>Prompted by an accident of a Boeing 747-131 (flight TWA800), the FAA published Special Federal Aviation Regulation (SFAR) 88, and the Joint Aviation Authorities (JAA) published Interim Policy INT/POL/25/12.</p> <p>The review conducted by Fokker Services on the Fokker 50/60 design in response to these regulations revealed that no controlled bonding provisions are present on a number of critical locations in the fuel system outside the fuel tanks.</p> <p>This condition, if not corrected, could create an ignition source in the fuel tank vapour space, possibly resulting in a fuel tank explosion and consequent loss of the aeroplane.</p> <p>For the reasons described above, this AD requires installation of additional electrical bonding provisions in the fuel system outside the fuel tanks and, subsequently, the implementation of the associated Fuel Airworthiness Limitation Item (ALI) and Critical Design Configuration Control Limitations (CDCCL).</p>	
Effective Date:	[TBD: 14 days after final AD issue date]	

<p>Required Action(s) and Compliance Time(s):</p>	<p>Required as indicated, unless accomplished previously.</p> <p>(1) Within 24 months after the effective date of this AD, install the additional bonding provisions in the fuel system outside the fuel tanks in accordance with the Accomplishment Instructions of Fokker Services Service Bulletin (SB) SBF50-28-039.</p> <p>Note 1: The actions required by paragraph (1) of this AD do not require opening of any fuel tank access panels.</p> <p>(2) Fuel ALI and CDCCL: After modification of an aeroplane as required by paragraph (1) of this AD, ensure that the additional bonding provisions remain installed on that aeroplane in compliance with the requirements of this AD.</p> <p>(3) Compliance with the requirement of paragraph (2) of this AD can be demonstrated by:</p> <p>(3.1) Revising as follows the approved aircraft maintenance programme on the basis of which the operator or the owner ensures the continuing airworthiness of each operated aeroplane:</p> <p style="padding-left: 40px;">Incorporate the Fuel ALI and CDCCL items in accordance with the information in paragraph 1.L.(1).(c) of Fokker Services SBF50-28-039,</p> <p style="padding-left: 40px;">and</p> <p>(3.2) Complying with the approved aircraft maintenance programme described in paragraph (3.1) of this AD.</p> <p>Note 2: More information on this subject can be found in Fokker Services All Operators Message AOF50.050#03.</p>
<p>Ref. Publications:</p>	<p>Fokker Services SBF50-28-039 dated 01 November 2012</p> <p>Fokker Services AOF50.050#03 dated 01 November 2012</p> <p>The use of later approved revisions of these documents is acceptable for compliance with the requirements of this AD.</p>
<p>Remarks:</p>	<p>1. This Proposed AD will be closed for consultation on 05 February 2013.</p> <p>2. Enquiries regarding this PAD should be referred to the Safety Information Section, Executive Directorate, EASA. E-mail: <a href="mailto:ADs@easa.europa.eu">ADs@easa.europa.eu</a>.</p> <p>3. For any question concerning the technical content of the requirements in this PAD, please contact: Fokker Services B.V., Technical Services Dept., P.O. Box 1357, 2130 EL, Hoofddorp, The Netherlands; telephone +31-88-6280-350; facsimile +31-88-6280-111; e-mail: <a href="mailto:technicalservices@fokker.com">technicalservices@fokker.com</a>. The referenced publication can be downloaded from <a href="http://www.myfokkerfleet.com">www.myfokkerfleet.com</a>.</p>