


EASA	NOTIFICATION OF A PROPOSAL TO ISSUE AN AIRWORTHINESS DIRECTIVE	
	PAD No.: 13-028 Date: 05 February 2013 Note: This Proposed Airworthiness Directive (PAD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.	
	In accordance with the EASA Continuing Airworthiness Procedures, the Executive Director is proposing the issuance of an EASA Airworthiness Directive (AD), applicable to the aeronautical product(s) identified below. All interested persons may send their comments, referencing the PAD Number above, to the e-mail address specified in the 'Remarks' section, prior to the consultation closing date indicated.	
Design Approval Holder's Name: AIRBUS		Type/Model designation(s): A318, A319, A320 and A321 aeroplanes
TCDS Number: EASA.A.064		
Foreign AD: Not applicable		
Supersedure: This AD supersedes EASA AD 2010-0071R1 dated 28 May 2010, and EASA AD 2011-0231 dated 09 December 2011.		
ATA 05		
Time Limits and Maintenance Checks – Damage Tolerant Airworthiness Limitation Items – ALS Part 2– Amendment		
Manufacturer(s): Airbus (formerly Airbus Industrie)		
Applicability: Airbus A318-111, A318-112, A318-121, A318-122, A319-111, A319-112, A319-113, A319-114, A319-115, A319-131, A319-132, A319-133, A320-111, A320-211, A320-212, A320-214, A320-215, A320-216, A320-231, A320-232, A320-233, A321-111, A321-112, A321-131, A321-211, A321-212, A321-213, A321-231 and A321-232 aeroplanes, all manufacturer serial numbers.		
Reason: <p>The airworthiness limitations for Airbus A320 family aeroplanes are currently included in Airbus A318/A319/A320/A321 Airworthiness Limitations Section (ALS) document. The airworthiness limitations applicable to the Damage Tolerant Airworthiness Limitation Items (DT ALI) are currently given in Airbus Airworthiness Limitations Section (ALS) Part 2, which is approved by EASA.</p> <p>Previously, EASA issued AD 2010-0071R1, which required the implementation of the DT ALI maintenance instructions as specified in Airbus A318/A319/A320-/A321 ALI Document ref.AI/SE-M4/95A.0252/96 issue 10 and Airbus A319 Corporate Jet ALI Document ref.AI/SE-M2/95A.1038/99.</p> <p>The new Airbus A318/A319/A320/A321 Airworthiness Limitations Section (ALS) Part 2 Revision 01 document which include also Airbus A319 Corporate Jet introduce more restrictive DT ALI maintenance instructions. Failure to comply with these instructions could result in an unsafe condition.</p> <p>Application of new DT ALI tasks 531129-02-2 and 531129-02-3 introduces initial and repetitive inspections of the windshield central lower node continuity fittings, replacing the one time inspection for that subject, previously required by</p>		

	<p>EASA AD 2011-0231.</p> <p>For the reasons described above, this AD retains the requirements of EASA AD 2010-0071R1 and those of EASA AD 2011-0231, which are superseded, and requires compliance with all maintenance tasks as described in Airbus A318/A319/A320/A321 ALS Part 2 at revision 01</p>
Effective Date:	[TBD: 14 days after final AD issue date]
Required Action(s) and Compliance Time(s):	<p>Required as indicated, unless accomplished previously:</p> <ol style="list-style-type: none"> (1) After the effective date of this AD, within the compliance times defined in the Record of Revisions (ROR) pages of the Airbus A318/A319/A320/A321 Airworthiness Limitations Section (ALS) Part 2 Revision 01 document, accomplish all applicable maintenance tasks, depending on aeroplane configuration, as specified Airbus A318/A319/A320/A321 Airworthiness Limitations Section (ALS) Part 2 Revision 01 document. (2) If, during accomplishment of any task as required by paragraph (1) of this AD, discrepancies are detected, within the applicable compliance time specified in the Airbus maintenance documentation, accomplish the applicable corrective action(s) in accordance with instructions specified (or incorporated by reference) in the applicable Airbus maintenance documentation. <p>If no compliance time is identified in the Airbus maintenance documentation, before next flight, accomplish the corrective action(s) or contact Airbus for approved instructions and within the compliance time specified in those instructions, accomplish those instructions accordingly.</p> <p>If any of the discrepancies detected during accomplishment of any task as required by paragraph (1) of this AD is not defined in the Airbus maintenance documentation, before next flight, contact Airbus for approved instructions and accomplish those instructions accordingly.</p> (3) Compliance with the requirements of paragraph (1) of this AD can be demonstrated by: <ol style="list-style-type: none"> (3.1) Revising as follows, the approved Aircraft Maintenance Programme (AMP) on the basis of which the operator or the owner ensures the continuing airworthiness of each operated aeroplane: <p>Incorporate all DT ALI maintenance instructions specified in Airbus A318/A319/A320/A321 Airworthiness Limitations Section (ALS) Part 2 Revision 01 document</p> <p>and</p> (3.2) Complying with the approved AMP described in paragraph (3.1) of this AD.
Ref. Publications:	<p>Airbus A318/A319/A320/A321 Airworthiness Limitations Section (ALS) Part 2 Revision 01 document approved by EASA on 04 April 2012.</p> <p>The use of later approved variations or revisions of this document is acceptable for compliance with the requirements of this AD.</p>
Remarks:	<ol style="list-style-type: none"> 1. This Proposed AD will be closed for consultation on 05 March 2013. 2. Enquiries regarding this PAD should be referred to the Safety Information Section, Executive Directorate, EASA. E-mail: ADs@easa.europa.eu. 3. For any question concerning the technical content of the requirements in this PAD, please contact: AIRBUS - Airworthiness Office - EIAS, Fax +33 5 61 93 44 51, E-mail: account.airworth-eas@airbus.com.