


<b>EASA</b>	<b>NOTIFICATION OF A PROPOSAL TO ISSUE AN AIRWORTHINESS DIRECTIVE</b>
	<p><b>PAD No.: 13-033</b></p> <p><b>Date: 06 February 2013</b></p> <p>Note: This Proposed Airworthiness Directive (PAD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.</p>
<p>In accordance with the EASA Continuing Airworthiness Procedures, the Executive Director is proposing the issuance of an EASA Airworthiness Directive (AD), applicable to the aeronautical product(s) identified below. All interested persons may send their comments, referencing the PAD Number above, to the e-mail address specified in the 'Remarks' section, prior to the consultation closing date indicated.</p>	
<b>Design Change Approval Holder's Name:</b> AIRBUS	<b>Type/Model designation(s):</b> A340 aeroplanes
TCDS Number:	EASA.A.015
Foreign AD:	Not applicable
Supersedure:	This AD supersedes EASA AD 2010-0036 dated 08 March 2010 and EASA AD 2008-0050 dated 04 March 2008.
<b>ATA 05</b>	<b>Time Limits and Maintenance Checks – Damage Tolerant Airworthiness Limitation Items – ALS Part 2 – Amendment</b>
Manufacturer(s):	Airbus (formerly Airbus Industrie)
Applicability:	Airbus A340-211, A340-212, A340-213, A340-311, A340-312, A340-313, A340-541, A340-542, A340-642 and A340-643 aeroplanes, all manufacturer serial numbers.
Reason:	<p>The airworthiness limitations are currently defined and published in the Airbus A340 Airworthiness Limitations Section (ALS).</p> <p>The airworthiness limitations applicable to the Damage Tolerant Airworthiness Limitation Items (DT ALI) were specified in Airbus A340 ALI Document reference AI/SE-M4/95A.0051/97, and currently in Airbus A340 ALS Part 2, which is approved by EASA.</p> <p>EASA issued AD 2010-0036 to cover Airbus ALI document issue 11.</p> <p>Since that AD was issued, more restrictive maintenance requirements and/or airworthiness limitations have been introduced in Revision 01 of the Airbus A340 ALS Part 2, including Fatigue and Damage Tolerance (F&amp;DT) evaluation and modification requirements, in order to prevent the occurrence of widespread fatigue damage (WFD) on the applicable aeroplanes.</p> <p>The modification requirements and their mandatory embodiment points (Structural Modification Point) are listed in a new Section 3.</p> <p>Failure to comply with the relevant instructions could result in an unsafe condition.</p> <p>In addition, in 2008, EASA issued AD 2008-0050 to require accomplishment of</p>

	<p>repetitive inspections of frame foot junction with lateral frame (ALI tasks 571123-01-01 and 571123-01-02). Those tasks have now been transferred in the A340 ALS Part 2 document.</p> <p>For the reasons described above, this AD retains the requirements of EASA AD 2010-0036 and AD 2008-0050, which are superseded, and requires the implementation of new or more restrictive maintenance instructions and/or airworthiness limitations as specified in Airbus A340 ALS Part 2 Revision 01. In addition, due to a mistake identified on the Table showing the WV Groups in paragraph 1.2.B of Section 1 of A340 ALS Part 2 Revision 01, the correct Table is published in Variation ref. 0FVLG120020/C0S which is required by this AD.</p>
Effective Date:	[TBD: 14 days after final AD issue date]
Required Action(s) and Compliance Time(s):	<p>Required as indicated, unless previously accomplished:</p> <p>(1) Except as specified in paragraph (2) of this AD, within the compliance times defined in the Record of Revisions (ROR) pages of the Airbus A340 ALS Part 2 Revision 01:</p> <p>Comply with all applicable instructions and airworthiness limitations included in sections 1 and 2 of Airbus A340 ALS Part 2 Revision 01, and the additional instructions specified in the Variation to Airbus A340 ALS Part 2 Revision 01, ref. 0FVLG120020/C0S.</p> <p>(2) Within the compliance times defined in Section 3 of Airbus A340 ALS Part 2 Revision 01, modify the aeroplane in accordance with the applicable instructions specified in Section 3 of Airbus A340 ALS Part 2 Revision 01.</p> <p>(3) Compliance with the requirements of paragraph (1) of this AD can be demonstrated by:</p> <p>(3.1) Revising as follows, the approved Aircraft Maintenance Programme (AMP) on the basis of which the operator or the owner ensures the continuing airworthiness of each operated aeroplane:</p> <p>Incorporate all maintenance requirements and associated airworthiness limitations specified in the Airbus A340 ALS Part 2 Revision 01 and its Variation ref. 0FVLG120020/C0S, as applicable to aeroplane model and weight variant,</p> <p>and</p> <p>(3.2) Complying with the approved AMP described in paragraph (3.1) of this AD.</p>
Ref. Publications:	<p>Airbus A340 ALS Part 2 Revision 01 approved by EASA on 05 October 2012, including Variation ref. 0FVLG120020/C0S approved by EASA on 23 November 2012.</p> <p>The use of later approved variations or revisions of this document is acceptable for compliance with the requirements of this AD.</p>
Remarks:	<ol style="list-style-type: none"> <li>1. This Proposed AD will be closed for consultation on 06 March 2013.</li> <li>2. Enquiries regarding this PAD should be referred to the Safety Information Section, Executive Directorate, EASA. E-mail: <a href="mailto:ADs@easa.europa.eu">ADs@easa.europa.eu</a>.</li> <li>3. For any question concerning the technical content of the requirements in this PAD, please contact: AIRBUS – Airworthiness Office – EIAL; E-mail: <a href="mailto:airworthiness.A330-A340@airbus.com">airworthiness.A330-A340@airbus.com</a>.</li> </ol>