

EASA	NOTIFICATION OF A PROPOSAL TO ISSUE AN AIRWORTHINESS DIRECTIVE
	<p>PAD No.: 13-065R1</p> <p>Date: 29 October 2014</p> <p>Note: This Proposed Airworthiness Directive (PAD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.</p>
<p>In accordance with the EASA Continuing Airworthiness Procedures, the Executive Director is proposing the issuance of an EASA Airworthiness Directive (AD), applicable to the aeronautical product(s) identified below. All interested persons may send their comments, referencing the PAD Number above, to the e-mail address specified in the 'Remarks' section, prior to the consultation closing date indicated.</p>	
Design Approval Holder's Name: AIRBUS	Type/Model designation(s): A318, A319, A320 and A321 aeroplanes
TCDS Number: EASA.A.064	
Foreign AD: Not applicable	
Supersedure: This AD supersedes EASA AD 2011-0155R1 dated 31 January 2012.	
ATA 05	Time Limits / Maintenance Checks – Airworthiness Limitations Section (ALS) Part 5 Fuel Airworthiness Limitations – Amendment
Manufacturer(s):	Airbus (formerly Airbus Industrie)
Applicability:	Airbus A318-111, A318-112, A318-121, A318-122, A319-111, A319-112, A319-113, A319-114, A319-115, A319-131, A319-132, A319-133, A320-211, A320-212, A320-214, A320-215, A320-216, A320-231, A320-232, A320-233, A321-111, A321-112, A321-131, A321-211, A321-212, A321-213, A321-231, and A321-232 aeroplanes, all manufacturer serial numbers
Reason:	<p>Prompted by an accident of a Boeing 747-131 (Flight TWA800); the Federal Aviation Administration (FAA) published Special Federal Aviation Regulation (SFAR) 88, and the Joint Aviation Authorities (JAA) published interim Policy INT/POL/25/12. In response to these regulations, Airbus conducted a design review to develop Fuel Airworthiness Limitations (FAL) for Airbus on A320 aeroplanes.</p> <p>The FAL were specified in Airbus A318/A319/A320/A321 FAL document ref. 95A.1931/05 at issue 04 for A318/A319/A320/A321 aeroplanes. This document was approved by the European Aviation Safety Agency (EASA) and is now referenced in Airbus A318/A319/A320/A321 ALS Part 5 to comply with EASA policy statement (EASA D2005/CPRO).</p> <p>Failure to comply with items as identified in Airbus A318/A319/A320/A321 ALS Part 5 could result in a fuel tank explosion and consequent loss of the aeroplane.</p> <p>Since PAD 13-065 was issued, Airbus policy regarding the incorporation of Fuel Tank Safety modifications into the ALS Part 5 has been amended, in agreement with EASA, as a result of which the modifications related to certain</p>

	<p>EASA and DGAC France ADs specified in the Supersedure section of the original PAD 13-065 have been removed from Airbus A318/A319/A320/A321 ALS Part 5 at Revision (Rev.) 01. Consequently, those ADs will not be superseded and remain active.</p> <p>For the reasons described above, this AD only retains the requirements of EASA AD 2011-0155R1, which is superseded, and requires implementation of the new or more restrictive maintenance requirements and/or airworthiness limitations as specified in Airbus A318/A319/A320/A321 ALS Part 5 at Rev.01.</p> <p>Note: Airbus A318/A319/A320/A321 ALS Part 5 at Rev.01 only introduces new or more restrictive tasks for aeroplanes operating under U.S. registry or regulations. Aeroplanes registered in an EASA Member State are not affected by the additional requirements, as compared to Airbus A318/A319/A320/A321 ALS Part 5 at Rev.00.</p> <p>This revised PAD has also been re-written (see RACT section below) to introduce the new EASA 'standard' for ALS ADs.</p>
Effective Date:	[TBD: 14 days after final AD issue date]
Required Action(s) and Compliance Time(s):	<p>Required as indicated, unless accomplished previously:</p> <ol style="list-style-type: none"> (1) From the effective date of this AD, accomplish the following actions, as specified in A318/A319/A320/A321 ALS Part 5 Rev.01, as applicable to aeroplane type and depending on aeroplane configuration: <ol style="list-style-type: none"> (1.1) Replace each component before or upon reaching the applicable life limit, and (1.2) Within the thresholds and intervals, accomplish all applicable maintenance tasks. (2) In case of finding discrepancies during accomplishment of any task as required by paragraph (1) of this AD, within the compliance time specified in the ALS Part 5 Rev.01, accomplish the applicable maintenance procedures for corrective actions in accordance with the approved maintenance documentation. If no compliance time is identified in the ALS Part 5 Rev.01, accomplish the applicable corrective actions before next flight. If a detected discrepancy is not identified in the ALS Part 5 Rev.01, before next flight, contact Airbus for approved instructions and accomplish those instructions accordingly. (3) Within 12 months after the effective date of this AD, revise the approved aircraft maintenance programme (AMP) and standard practices, on the basis of which the operator or the owner ensures the continuing airworthiness of each operated aeroplane, by incorporating the limitations, tasks and associated thresholds and intervals described in the ALS Part 5 Rev.01. (4) For an AMP that, on the effective date of this AD, is already updated to incorporate the maintenance tasks and life limitations as specified in the ALS Part 5 at Rev.00, the new and more restrictive tasks and limitations, as defined in the ALS Part 5 at Rev.01, must be incorporated into the AMP to comply with paragraph (3) of this AD. (5) Compliance with the requirements of paragraph (3) constitutes compliance with the requirements of paragraphs (1) and (2) of this AD. After revising the AMP, as required by paragraph (3) or (4) of this AD, as applicable, it is not necessary that accomplishment of individual tasks is recorded for demonstration of AD compliance on a continued basis. <p>Note: For affected Airbus A318/A319/A320/A321 registered in Europe, complying with the approved AMP as specified in paragraph (3) of this AD is required by Commission Regulation (EC) No 2042/2003, Part M.A.301, paragraph 3.</p>

Ref. Publications:	<p>Airbus A318/A319/A320/A321 ALS Part 5 Rev.01, approved by EASA on 09 July 2014</p> <p>The use of later approved variations or revisions of this document is acceptable for compliance with the requirements of this AD.</p>
Remarks:	<ol style="list-style-type: none"> 1. This Proposed AD will be closed for consultation on 26 November 2014. 2. Enquiries regarding this AD should be referred to the Safety Information Section, Certification Directorate, EASA. E-mail: ADs@easa.europa.eu. 3. For any question concerning the technical content of the requirements in this AD, please contact: AIRBUS – Airworthiness Office – EIAS; Fax +33 5 61 93 44 51; E-mail: account.airworth-eas@airbus.com.