


EASA	AIRWORTHINESS DIRECTIVE
	<p>AD No.: 2014-0260</p> <p>Date: 05 December 2014</p> <p>Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.</p>
<p>This AD is issued in accordance with EU 748/2012, Part 21.A.3B. In accordance with EC 2042/2003 Annex I, Part M.A.301, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [EC 2042/2003 Annex I, Part M.A.303] or agreed with the Authority of the State of Registry [EC 216/2008, Article 14(4) exemption].</p>	
<p>Design Approval Holder's Name:</p> <p>AIRBUS</p>	<p>Type/Model designation(s):</p> <p>A318, A319, A320 and A321 aeroplanes</p>
TCDS Number:	EASA.A.064
Foreign AD:	Not applicable
Supersedure:	This AD supersedes EASA AD 2011-0155R1 dated 31 January 2012.
ATA 05	Time Limits / Maintenance Checks – Airworthiness Limitations Section (ALS) Part 5 Fuel Airworthiness Limitations – Amendment
Manufacturer(s):	Airbus (formerly Airbus Industrie)
Applicability:	Airbus A318-111, A318-112, A318-121, A318-122, A319-111, A319-112, A319-113, A319-114, A319-115, A319-131, A319-132, A319-133, A320-211, A320-212, A320-214, A320-215, A320-216, A320-231, A320-232, A320-233, A321-111, A321-112, A321-131, A321-211, A321-212, A321-213, A321-231, and A321-232 aeroplanes, all manufacturer serial numbers.
Reason:	<p>Prompted by an accident of a Boeing 747-131 (Flight TWA800), the Federal Aviation Administration (FAA) published Special Federal Aviation Regulation (SFAR) 88, and the Joint Aviation Authorities (JAA) published interim Policy INT/POL/25/12. In response to these regulations, Airbus conducted a design review to develop Fuel Airworthiness Limitations (FAL) for Airbus A320 family aeroplanes.</p> <p>The FAL were specified in Airbus A318/A319/A320/A321 FAL document ref. 95A.1931/05 at issue 04 for A318/A319/A320/A321 aeroplanes. This document was approved by the European Aviation Safety Agency (EASA) and is now referenced in Airbus A318/A319/A320/A321 ALS Part 5 to comply with EASA policy statement (EASA D2005/CPRO).</p> <p>Failure to comply with items as identified in Airbus A318/A319/A320/A321 ALS Part 5 could result in a fuel tank explosion and consequent loss of the aeroplane.</p> <p>For the reasons described above, this AD retains the requirements of EASA AD 2011-0155R1, which is superseded, and requires implementation of the new or more restrictive maintenance requirements and/or airworthiness limitations as specified in Airbus A318/A319/A320/A321 ALS Part 5 at Rev.01.</p>

	<p>Note: Airbus A318/A319/A320/A321 ALS Part 5 at Rev.01 only introduces new or more restrictive tasks for aeroplanes operating under U.S. registry or regulations. Aeroplanes registered in an EASA Member State are not affected by the additional requirements, as compared to Airbus A318/A319/A320/A321 ALS Part 5 at Rev.00.</p>
Effective Date:	19 December 2014
Required Action(s) and Compliance Time(s):	<p>Required as indicated, unless accomplished previously:</p> <ol style="list-style-type: none"> (1) From the effective date of this AD, accomplish the following actions, as specified in A318/A319/A320/A321 ALS Part 5 Rev.01, as applicable to aeroplane type and depending on aeroplane configuration: <ol style="list-style-type: none"> (1.1) Replace each component before or upon reaching the applicable life limit, and (1.2) Within the thresholds and intervals, accomplish all applicable maintenance tasks. (2) In case of finding discrepancies during accomplishment of any task as required by paragraph (1) of this AD, within the compliance time specified in the ALS Part 5 Rev.01, accomplish the applicable maintenance procedures for corrective actions in accordance with the approved maintenance documentation. If no compliance time is identified in the ALS Part 5 Rev.01, accomplish the applicable corrective actions before next flight. If a detected discrepancy is not identified in the ALS Part 5 Rev.01, before next flight, contact Airbus for approved instructions and accomplish those instructions accordingly. (3) Within 12 months after the effective date of this AD, revise the approved aircraft maintenance programme (AMP), on the basis of which the operator or the owner ensures the continuing airworthiness of each operated aeroplane, by incorporating the limitations, tasks and associated thresholds and intervals described in the ALS Part 5 Rev.01. (4) For an AMP that, on the effective date of this AD, is already updated to incorporate the maintenance tasks and life limitations as specified in the ALS Part 5 at Rev.00, the new and more restrictive tasks and limitations, as defined in the ALS Part 5 at Rev.01, must be incorporated into the AMP to comply with paragraph (3) of this AD. (5) Compliance with the requirements of paragraph (3) constitutes compliance with the requirements of paragraphs (1) and (2) of this AD. After revising the AMP, as required by paragraph (3) or (4) of this AD, as applicable, it is not necessary that accomplishment of individual tasks is recorded for demonstration of AD compliance on a continued basis. <p>Note: For affected Airbus A318/A319/A320/A321 registered in Europe, complying with the approved AMP as specified in paragraph (3) of this AD is required by Commission Regulation (EC) No 2042/2003, Part M.A.301, paragraph 3.</p>
Ref. Publications:	<p>Airbus A318/A319/A320/A321 ALS Part 5 Rev.01, approved by EASA on 09 July 2014.</p> <p>The use of later approved variations or revisions of this document is acceptable for compliance with the requirements of this AD.</p>
Remarks:	<ol style="list-style-type: none"> 1. If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD. 2. This AD was posted on 22 May 2013 as PAD 13-065 for consultation until 19 June 2013 and republished as PAD 13-065R1 on 29 October 2014 for consultation until 26 November 2014. The Comment Response Document can be found at http://ad.easa.europa.eu/.

	<ol style="list-style-type: none">3. Enquiries regarding this AD should be referred to the Safety Information Section, Certification Directorate, EASA. E-mail: ADs@easa.europa.eu.4. For any question concerning the technical content of the requirements in this AD, please contact: AIRBUS – Airworthiness Office – EIAS; Fax +33 5 61 93 44 51; E-mail: account.airworth-eas@airbus.com.
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