

<b>EASA</b>	<b>AIRWORTHINESS DIRECTIVE</b>
	<p><b>AD No.: 2013-0248</b></p> <p><b>Date: 14 October 2013</b></p> <p>Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.</p>
<p>This AD is issued in accordance with EU 748/2012, Part 21.A.3B. In accordance with EC 2042/2003 Annex I, Part M.A.301, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [EC 2042/2003 Annex I, Part M.A.303] or agreed with the Authority of the State of Registry [EC 216/2008, Article 14(4) exemption].</p>	
<p><b>Design Approval Holder's Name:</b> AIRBUS</p>	<p><b>Type/Model designation(s):</b> A300, A300-600, A300-600ST and A310 aeroplanes</p>
TCDS Number:	France n° 145 and EASA.A.014
Foreign AD:	Not applicable
Supersedure:	This AD supersedes EASA AD 2007-0293 dated 29 November 2007.
<b>ATA 05</b>	<b>Time limits / Maintenance checks – Safe life Airworthiness Limitation Items (ALS Part 1) – Amendment</b>
Manufacturer(s):	Airbus (formerly Airbus Industrie)
Applicability:	<p>Airbus A300 aeroplanes, all certified models, all Manufacturer Serial Numbers (MSN),</p> <p>Airbus A300-600 aeroplanes, all certified models, all MSN,</p> <p>Airbus A300F4-608ST aeroplanes, all MSN,</p> <p>Airbus A310 aeroplanes, all certified models all MSN.</p>
Reason:	<p>The airworthiness limitations are currently distributed in the Airbus A300 Airworthiness Limitation Section (ALS), A300-600 ALS and A310 ALS.</p> <p>The instructions and airworthiness limitations applicable to the Safe Life Airworthiness Limitation Items (SL ALI) are given in Airbus A300 ALS Part 1, A300-600 ALS Part 1 and A310 ALS Part 1, which are approved by the European Aviation Safety Agency (EASA).</p> <p>The revision 01 of Airbus A300 ALS Part 1, A300-600 ALS Part 1 and A310 ALS Part 1 introduces more restrictive instructions and/or airworthiness limitations. Failure to comply with the maintenance requirements contained in these revisions could constitute an unsafe condition.</p> <p>For the reasons described above, this AD retains the requirements of EASA AD 2007-0293, which is superseded, and requires the implementation of the new or more restrictive maintenance requirements and/or airworthiness limitations as specified in the Airbus A300 ALS Part 1 Revision 1, A300-600 ALS Part 1 Revision 1 and A310 ALS Part 1 Revision 1, except for the A300 life limits</p>

	applicable to Main Landing Gear (MLG) barrel assembly, retraction actuator assembly linkage assembly and to flanged duct, for which applicable life limits are defined in EASA AD 2013-0210.
Effective Date:	28 October 2013
Required Action(s) and Compliance Time(s):	<p>Required as indicated, unless accomplished previously:</p> <p>(1) From the effective date of this AD, accomplish the following actions, as specified in the Airbus A300 ALS Part 1 revision 01, Airbus A300-600 ALS Part 1 revision 01, or Airbus A310 ALS Part 1 revision 01, as applicable to aeroplane model and depending on aeroplane configuration:</p> <p>(1.1) Replace each component before or upon reaching the applicable life limit.</p> <p>(1.2) Within the thresholds and intervals, accomplish all applicable maintenance tasks.</p> <p>(2) Compliance with the requirements of paragraph (1) of this AD can be demonstrated by:</p> <p>(2.1) Revising as follows, the approved Aircraft Maintenance Program (AMP) on the basis of which the Operator or the Owner ensures the continuing airworthiness of each operated aeroplane:</p> <p>Incorporate all maintenance tasks and applicable airworthiness limitations specified in the Airbus A300 ALS Part 1 revision 01, Airbus A300-600 ALS Part 1 revision 01 and Airbus A310 ALS Part 1 revision 01, as applicable to aeroplane model and relevant to the aeroplane configuration and weight variant,</p> <p>and</p> <p>(2.2) Complying with the approved AMP described in paragraph (2.1) of this AD.</p>
Ref. Publications:	<p>Airbus A300 ALS Part 1 Revision 01 approved by EASA on 05 September 2013.</p> <p>Airbus A300-600 ALS Part 1 Revision 01 approved by EASA on 05 September 2013.</p> <p>Airbus A310 ALS Part 1 Revision 01 approved by EASA on 05 September 2013.</p> <p>The use of later approved variations or revisions of these documents is acceptable for compliance with the requirements of this AD.</p>
Remarks:	<ol style="list-style-type: none"> <li>If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.</li> <li>This AD was posted on 11 September 2013 as PAD 13-137 for consultation until 02 October 2013. No comments were received during the consultation period.</li> <li>Enquiries regarding this AD should be referred to the Safety Information Section, Executive Directorate, EASA. E-mail: <a href="mailto:ADs@easa.europa.eu">ADs@easa.europa.eu</a>.</li> <li>For any question concerning the technical content of the requirements in this AD, please contact: AIRBUS SAS – EIAW (Airworthiness Office), Telephone: + 33 5 61 18 41 39, Fax: + 33 5 61 93 44 51.</li> </ol>