


EASA	NOTIFICATION OF A PROPOSAL TO ISSUE AN AIRWORTHINESS DIRECTIVE
	<p>PAD No.: 14-120</p> <p>Date: 23 July 2014</p> <p>Note: This Proposed Airworthiness Directive (PAD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.</p>
<p>In accordance with the EASA Continuing Airworthiness Procedures, the Executive Director is proposing the issuance of an EASA Airworthiness Directive (AD), applicable to the aeronautical product(s) identified below. All interested persons may send their comments, referencing the PAD Number above, to the e-mail address specified in the 'Remarks' section, prior to the consultation closing date indicated.</p>	
<p>Design Approval Holder's Name:</p> <p>AIRBUS</p>	<p>Type/Model designation(s):</p> <p>A310 aeroplanes</p>
<p>TCDS Number: EASA.A.172</p>	
<p>Foreign AD: Not applicable</p>	
<p>Supersedure: None</p>	
ATA 71	Powerplant – Side Link Clevis Engine Mount Bolts – Replacement
Manufacturer(s):	Airbus (formerly Airbus Industrie)
Applicability:	Airbus A310-203 aeroplanes, all manufacturer serial numbers
Reason:	<p>During fatigue analysis performed in the scope of the Extended Service Goal, taking into account the certification loads and the new lift-off loads, Airbus determined that side link clevis engine mount bolts do not meet the Design Service Goal (DSG) requirements on aeroplanes equipped with CF6-80A3 engines.</p> <p>This condition, if not corrected, could lead to failure of the front engine mount, possibly resulting in-flight separation of the engine from the aeroplane.</p> <p>To address this potential unsafe condition, Airbus issued Service Bulletin (SB) A310-71-2038 to introduce a life limit on the side link clevis engine mount bolts.</p> <p>For the reason described above, this AD requires implementation of the new life limit and replacement of all side link clevis engine mount bolts that have exceeded the new limit.</p>
Effective Date:	[TBD: 14 days after final AD issue date]

Required Action(s) and Compliance Time(s):	<p>Required as indicated, unless accomplished previously:</p> <p>Within 18 months after the effective date of this AD, and, thereafter, at intervals not to exceed 29 years, replace the side link clevis engine mount bolts, nuts and bushings of the front engine mount on both engines and re-identify the new installed bolts with a cross (to differentiate them from the old ones) in accordance with the instructions of Airbus SB A310-71-2038.</p>
Ref. Publications:	<p>Airbus SB A310-71-2038 original issue dated 08 April 2014.</p> <p>The use of later approved revisions of this document is acceptable for compliance with the requirements of this AD.</p>
Remarks:	<ol style="list-style-type: none"> 1. This Proposed AD will be closed for consultation on 20 August 2014. 2. Enquiries regarding this PAD should be referred to the Safety Information Section, Executive Directorate, EASA. E-mail: ADs@easa.europa.eu. 3. For any question concerning the technical content of the requirements in this PAD, please contact: AIRBUS SAS – EIAW (Airworthiness Office) E-mail: continued.airworthiness-wb.external@airbus.com.